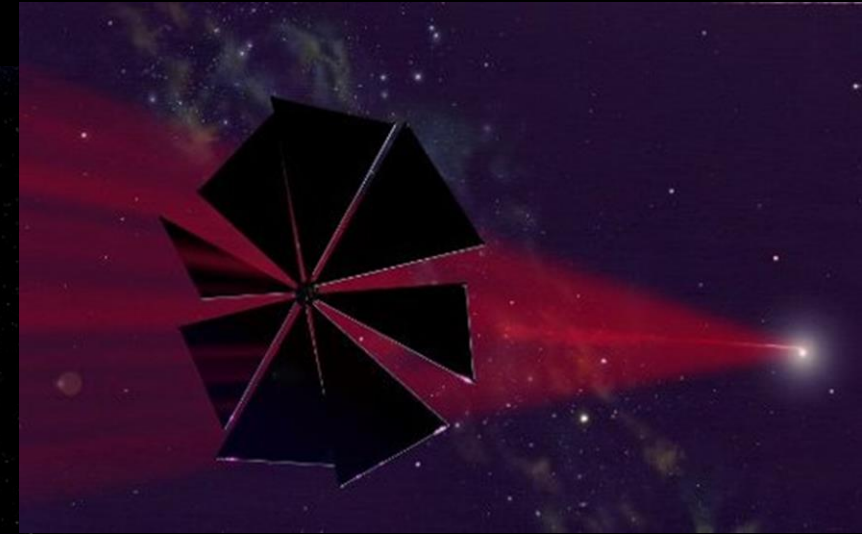
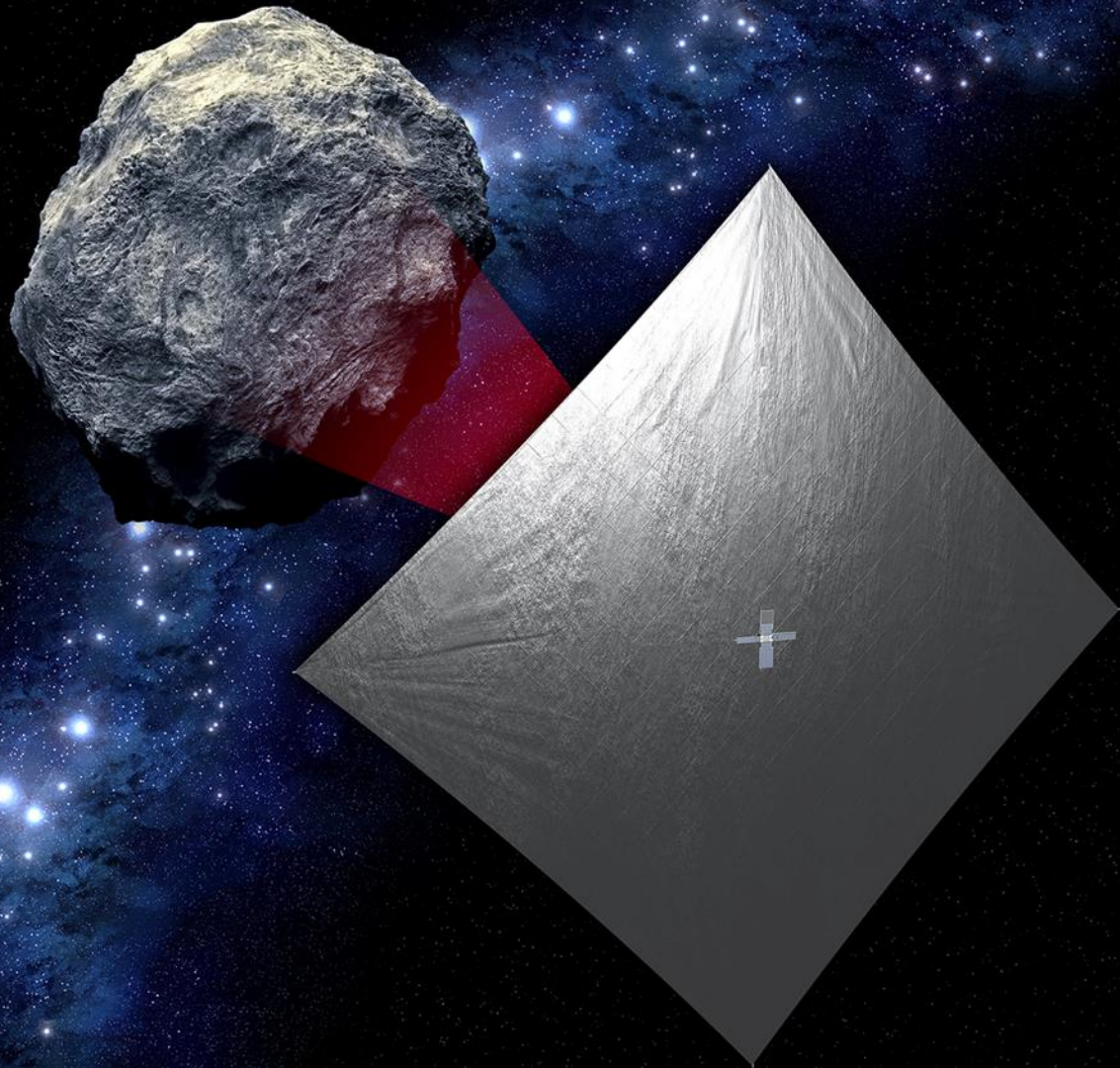




Solar Sails for Spacecraft Propulsion

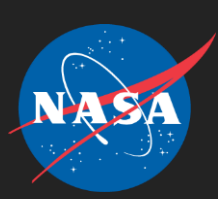


Les Johnson
NASA George C.
Marshall Space
Flight Center

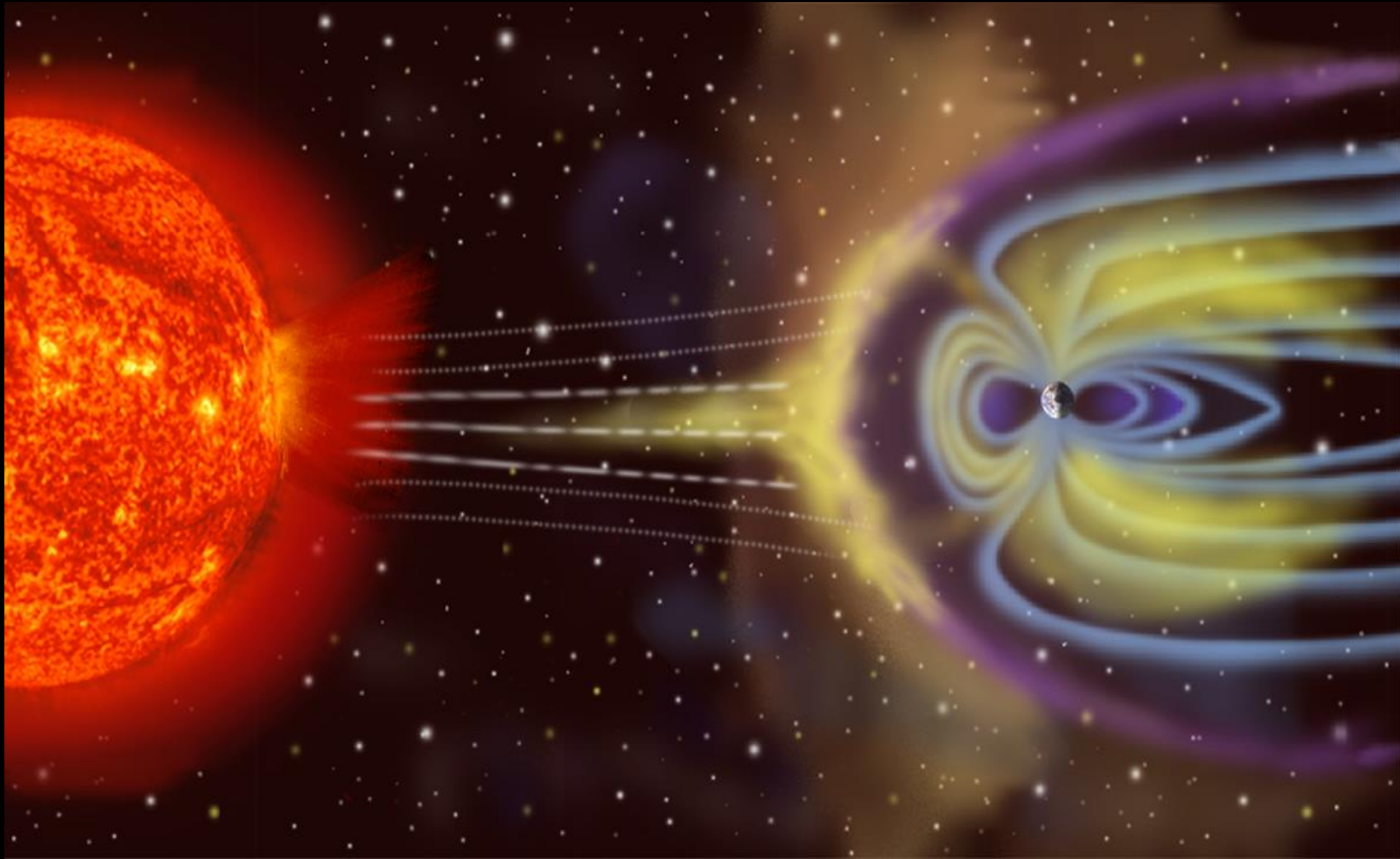


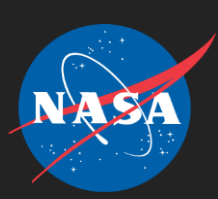
We tend to think of space as being big and empty...





Can we use the environments of space to our advantage?

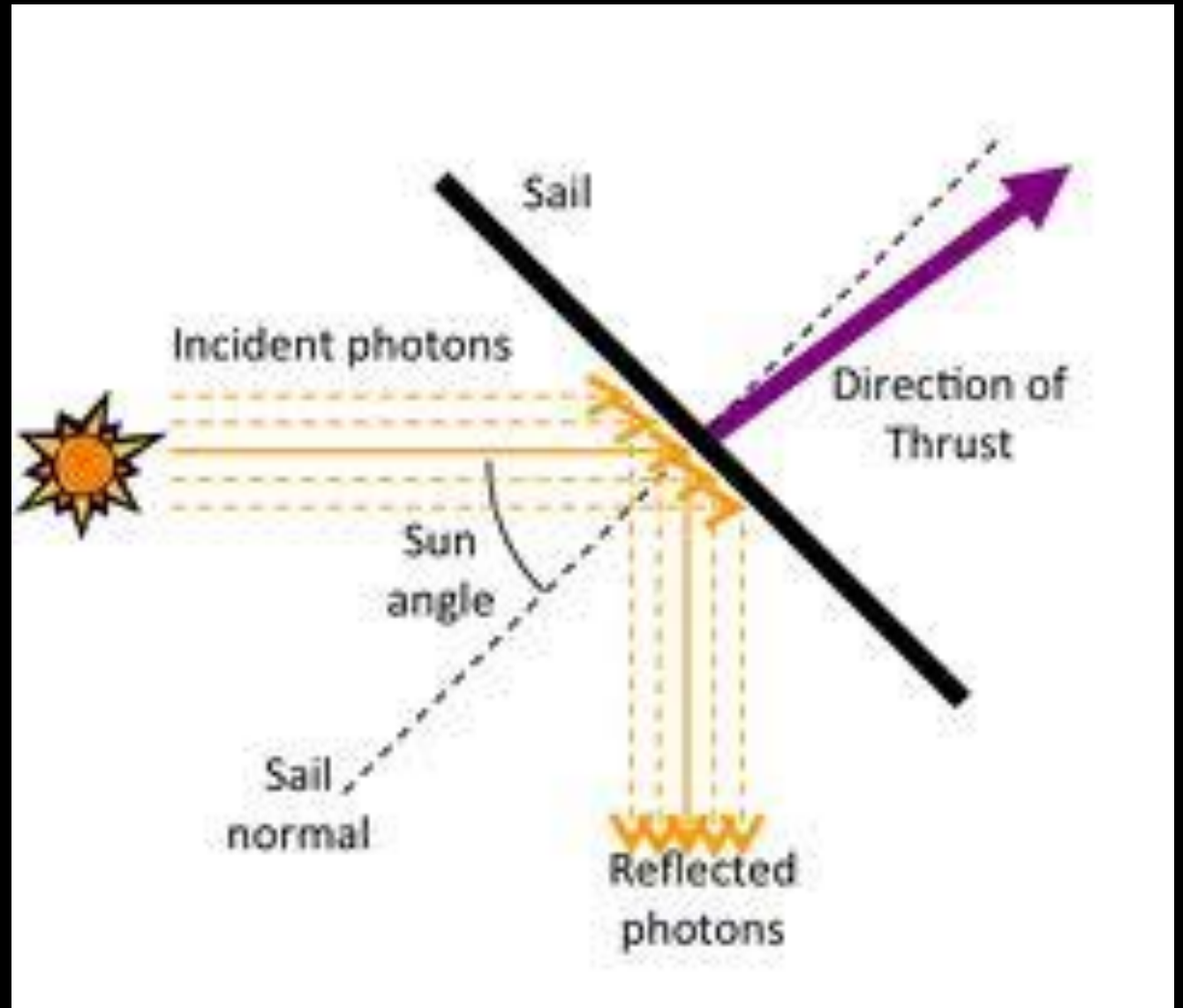
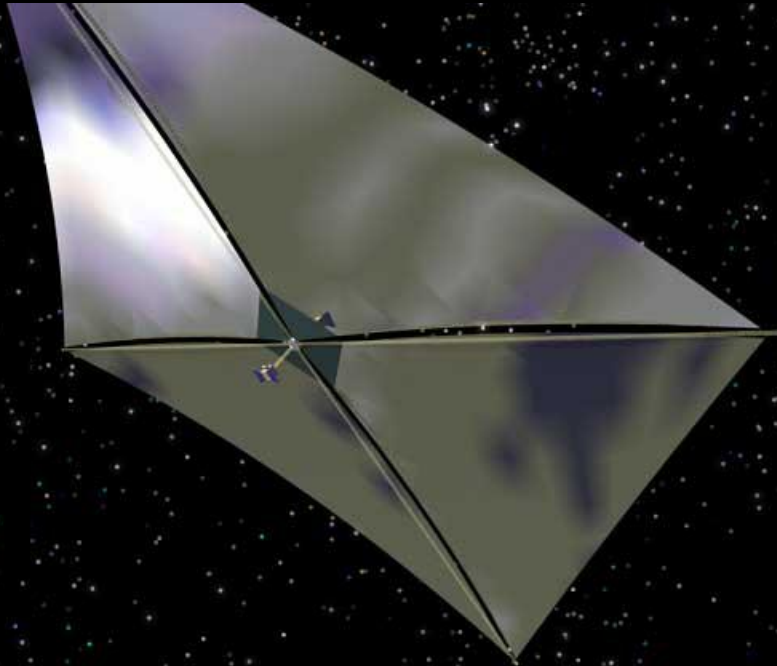




How does a solar sail work?



Solar sails use photon “pressure” or force on thin, lightweight reflective sheet to produce thrust.



- **Photons carry Momentum**

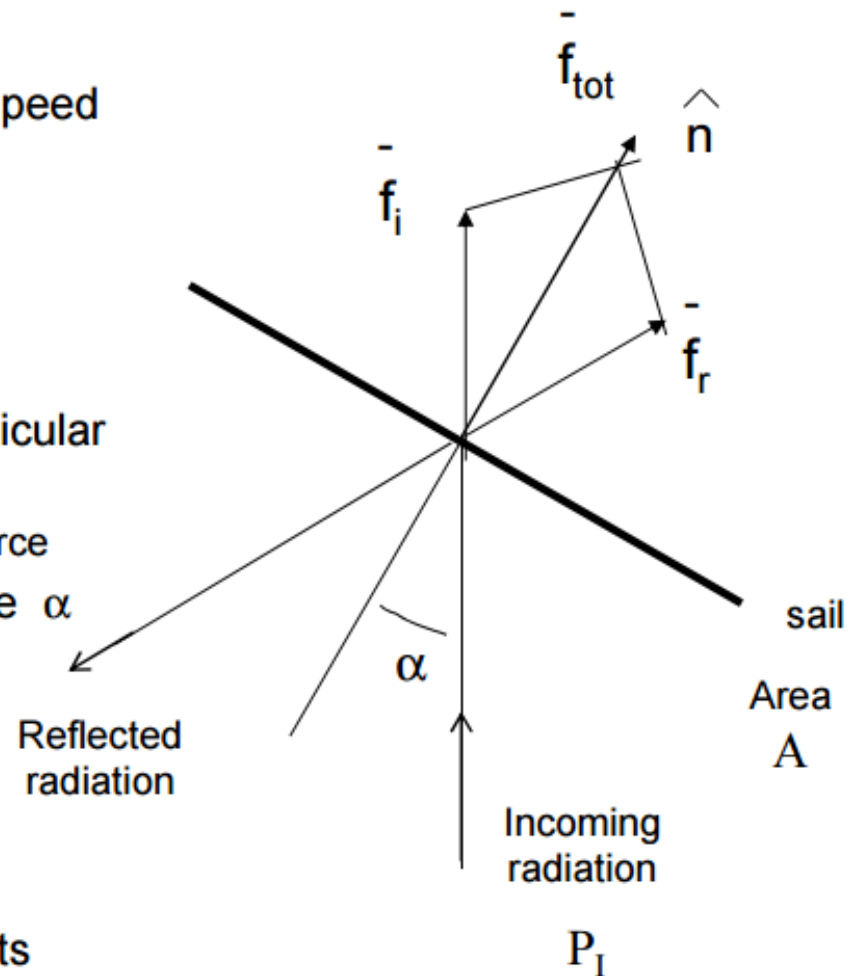
- $p = hv/c$
 - h = Planck's, ν = frequency, c = speed of light

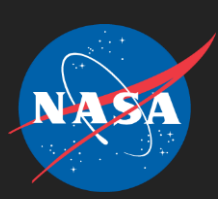
- **Force generated on Reflective Surface**

- Resultant force approximately perpendicular to surface
 - The bigger the surface, the more the force
- Can “steer” sail by changing pitch angle α

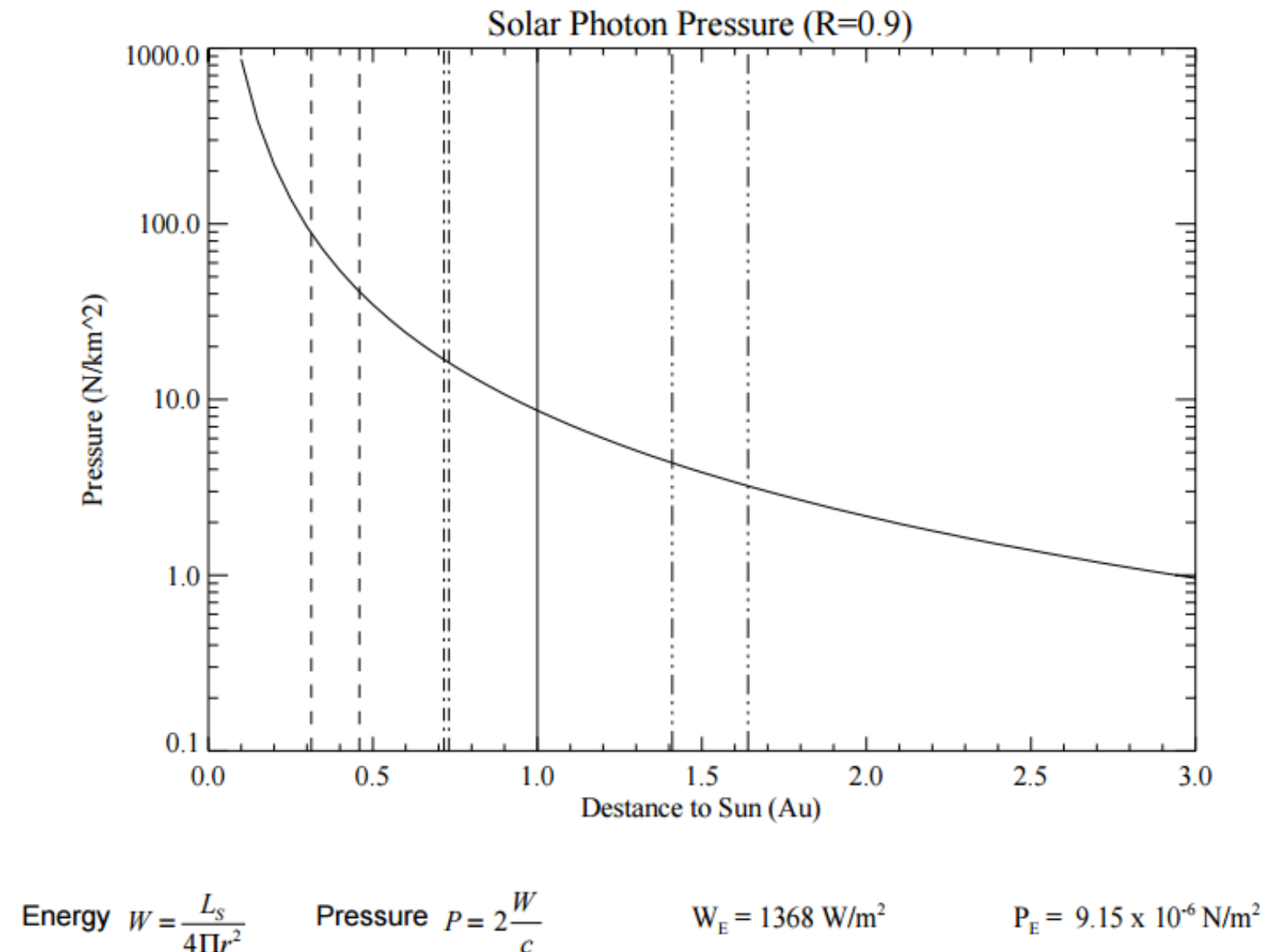
- **Small, but potentially Constant Acceleration**

- Potentially unlimited “delta V”
- Allows some otherwise impossible orbits

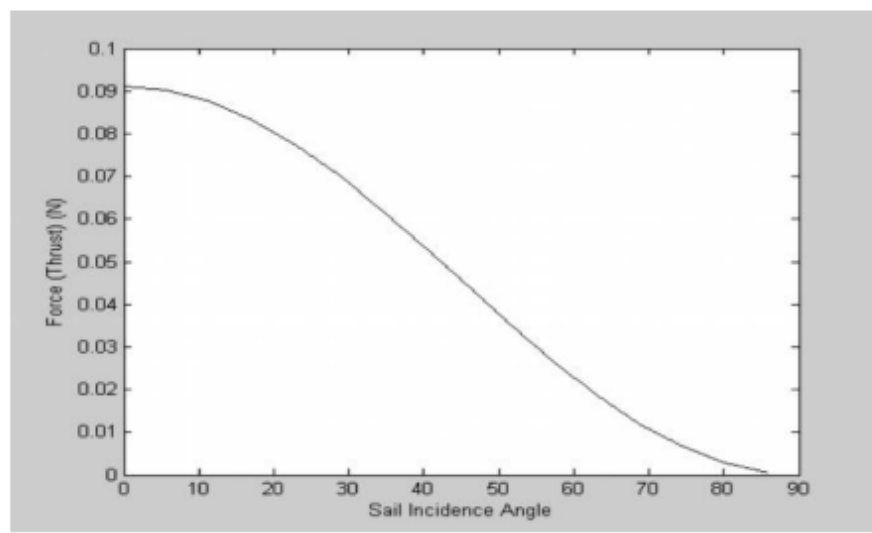




Solar Sails Provide Low Thrust Propulsion



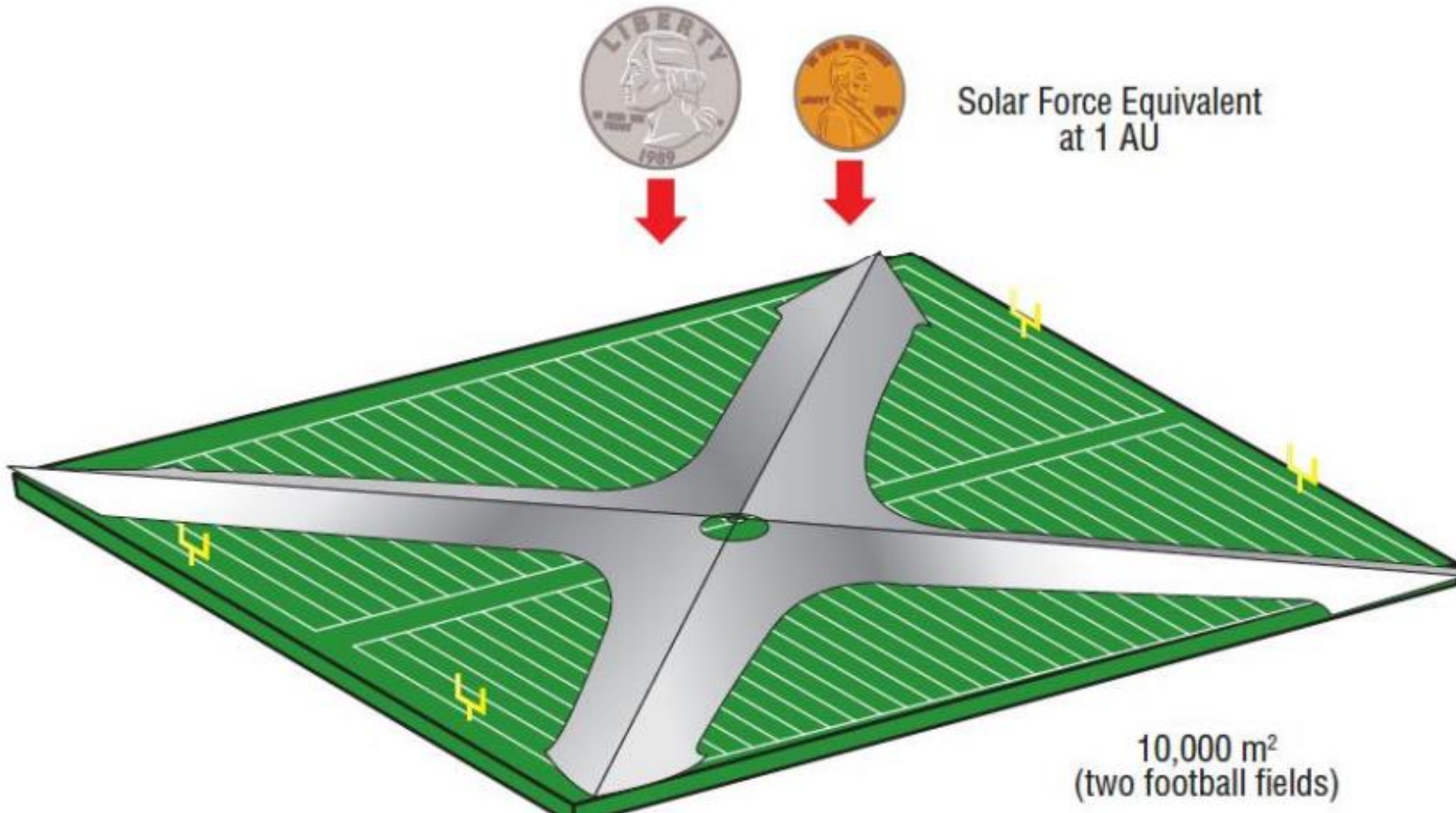
- For a 100 kg sailcraft, 100 m x 100 m square sail:
- Force (maximum, perp. to sun, perfect/flat reflector)
 - 0.09 N
- Acceleration (maximum)
 - $0.92 \times 10^{-3} \text{ m/s}^2$ (0.9 millimeters/sec²)
- Force decreases with increasing pitch angle (θ_i or α)

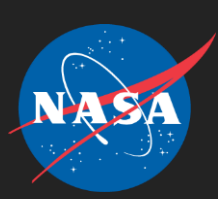


$$\bar{f}_{tot} = 2P_I A (\cos\theta_i)^2 \hat{n}$$

Solar Sail Total Force (Thrust) Vs. Sun-Incidence Angle
(For a 100 x 100 meter perfect sail @ 1 A.U.)

- Force on a 100 m x 100 m square sail:

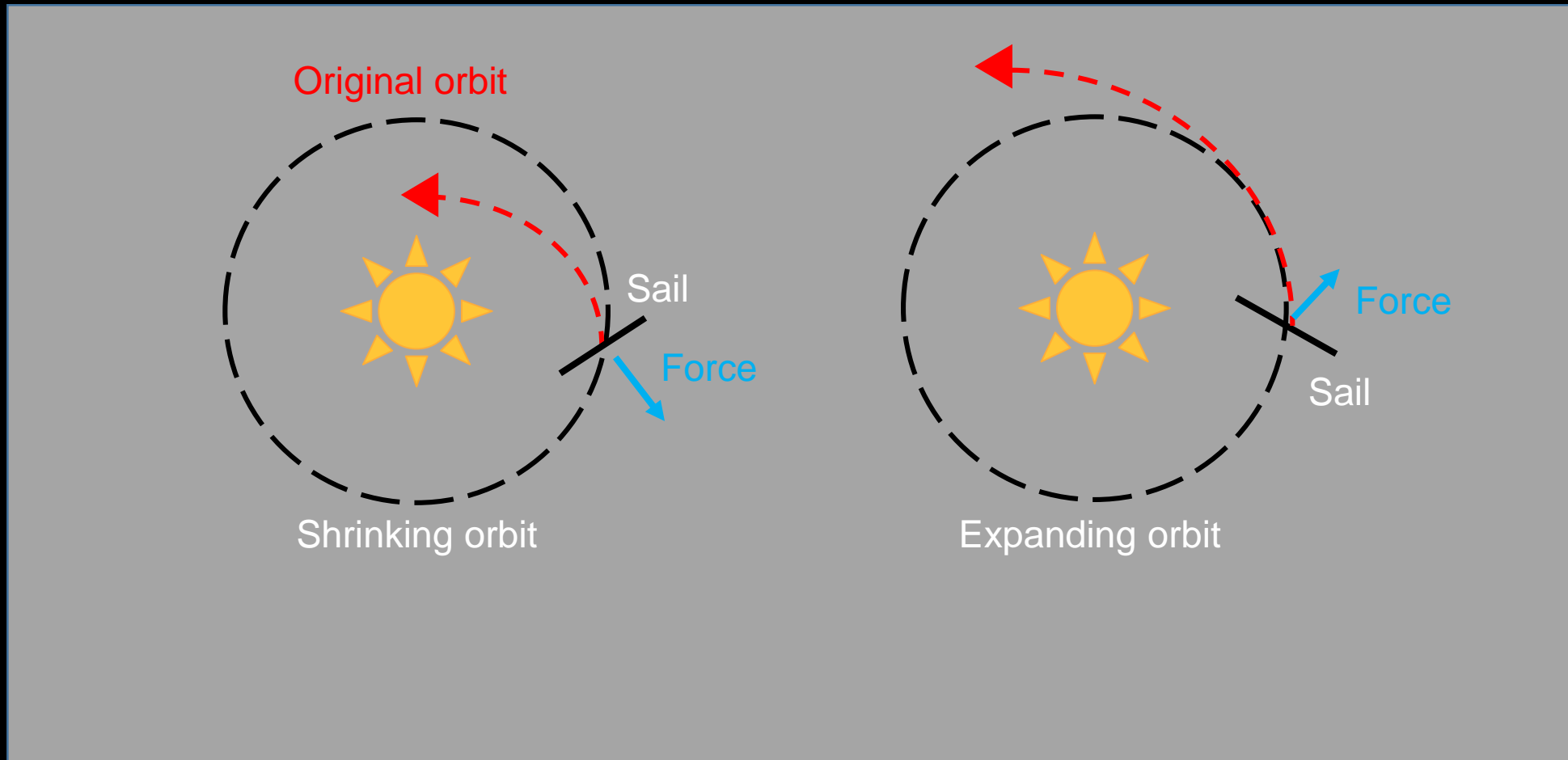


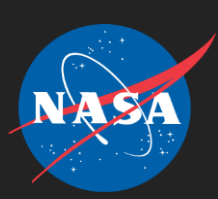


Solar Sail Trajectory Control



- Solar Radiation Pressure allows inward or outward Spiral





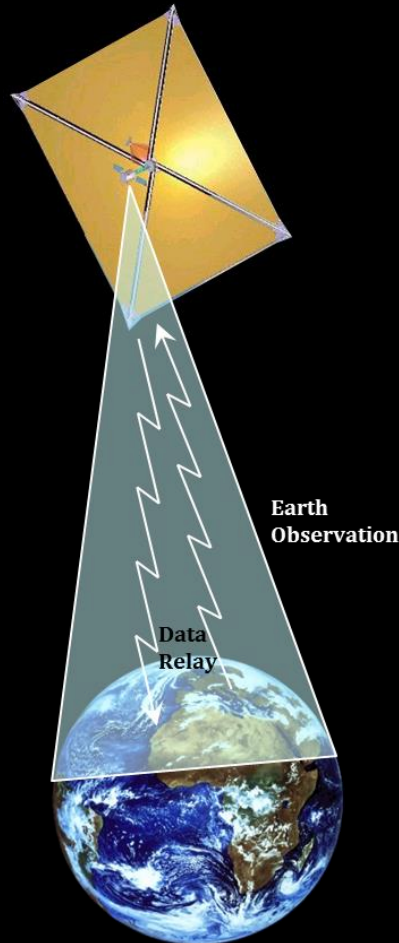
Potential Solar Sail Applications (A Partial List!)



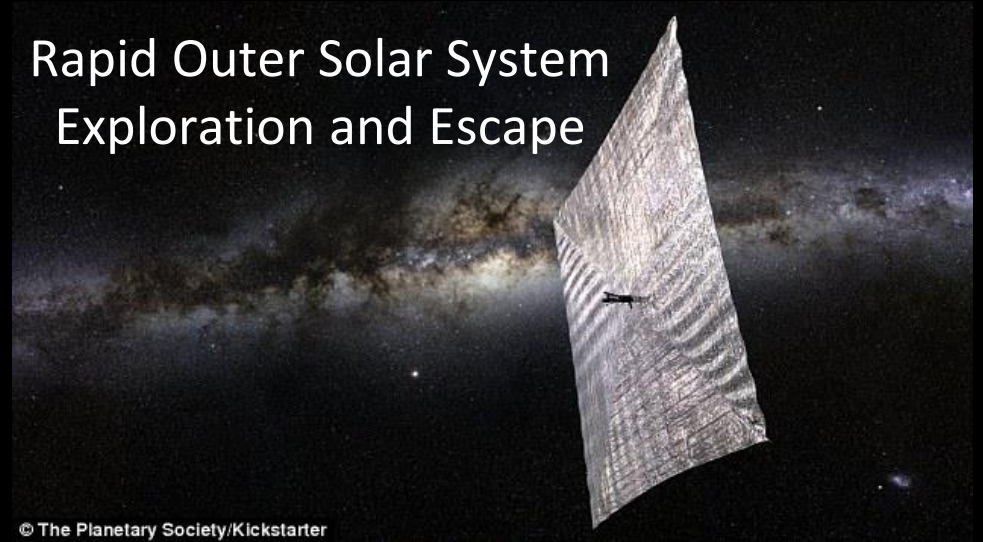
NEA Reconnaissance &
Small Body Science



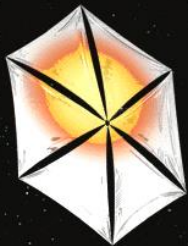
Earth Pole Sitting



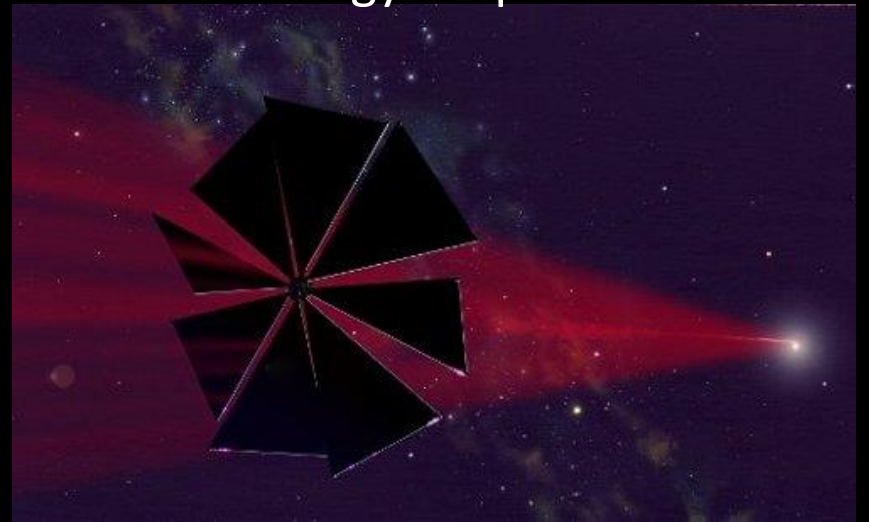
Rapid Outer Solar System
Exploration and Escape



Heliophysics & Out of the
Ecliptic Science



Toward Higher Performance Beamed
Energy Propulsion



Echo II 1964

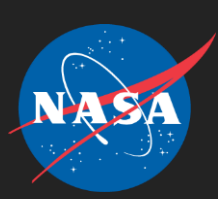
Solar thrust effect on spacecraft orbit



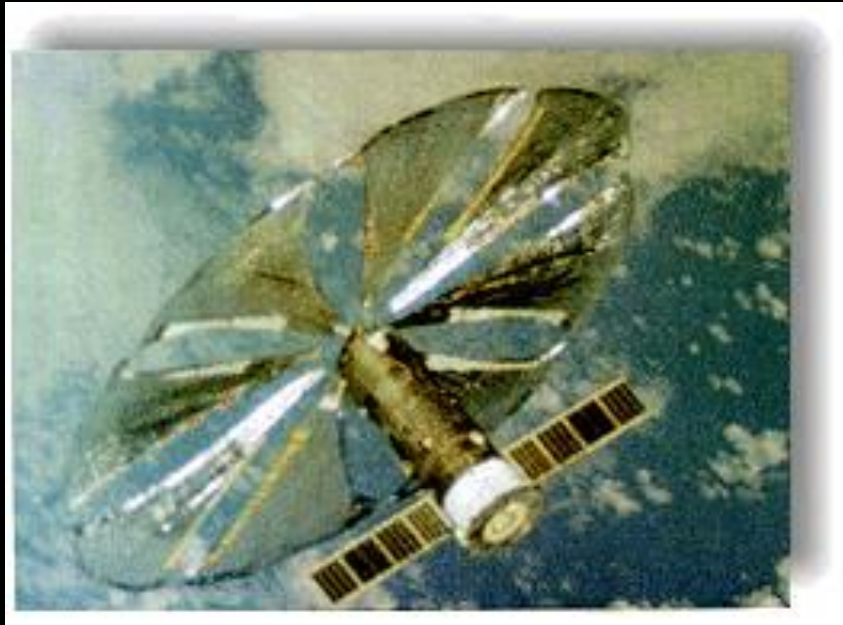
When folded, satellite was packed into the 41-inch diameter canister shown in the foreground.

- 135-foot rigidized inflatable balloon satellite
- laminated Mylar plastic and aluminum
- placed in near-polar Orbit
- passive communications experiment by NASA on January 25, 1964

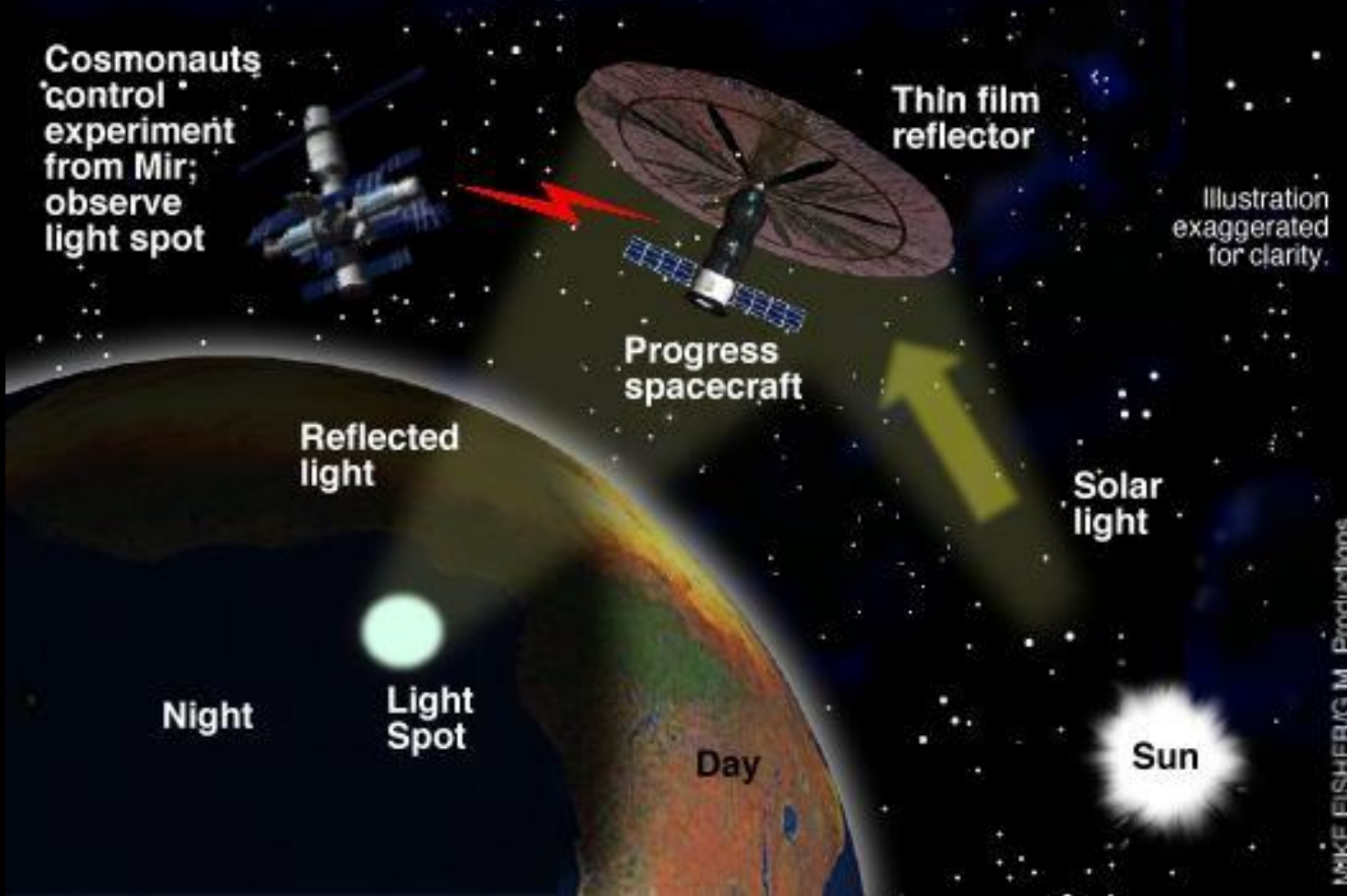




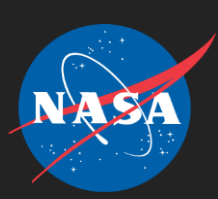
Znamya (Space Mirror)



“Novey Svet” (New Light) Experiment — Znamya 2.5



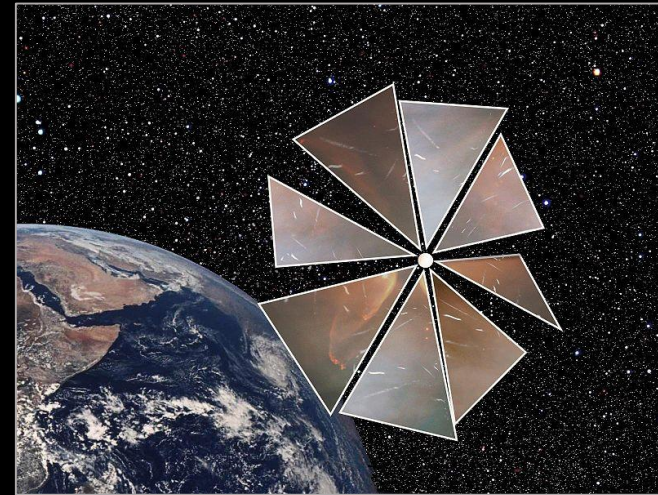
MIKE FISHER/G.M. Productions

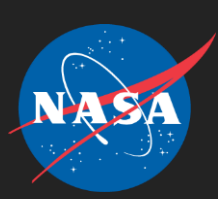


The Planetary Society's Cosmos-1 (2005)



- 100 kg spacecraft
- 8 triangular sail blades deployed from a central hub after launch by the inflating of structural tubes.
 - Sail blades were each 15 m long
 - Total surface area of 600 m²
- Launched in 2005 from a Russian Volna Rocket from a Russian Delta III submarine in the Barents Sea:

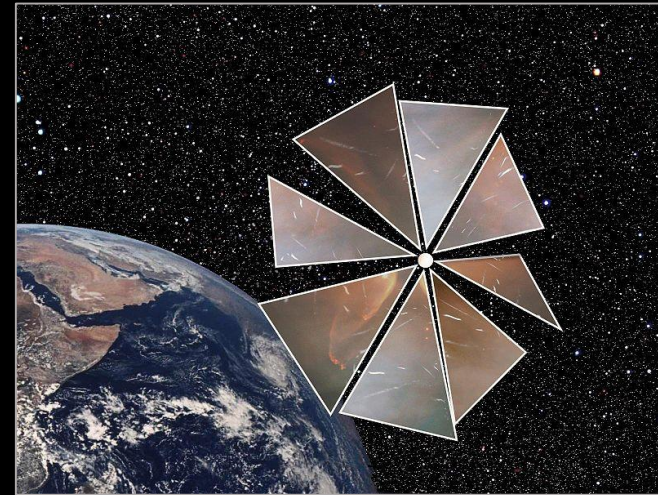




The Planetary Society's Cosmos-1 (2005)

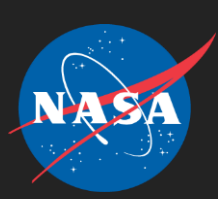


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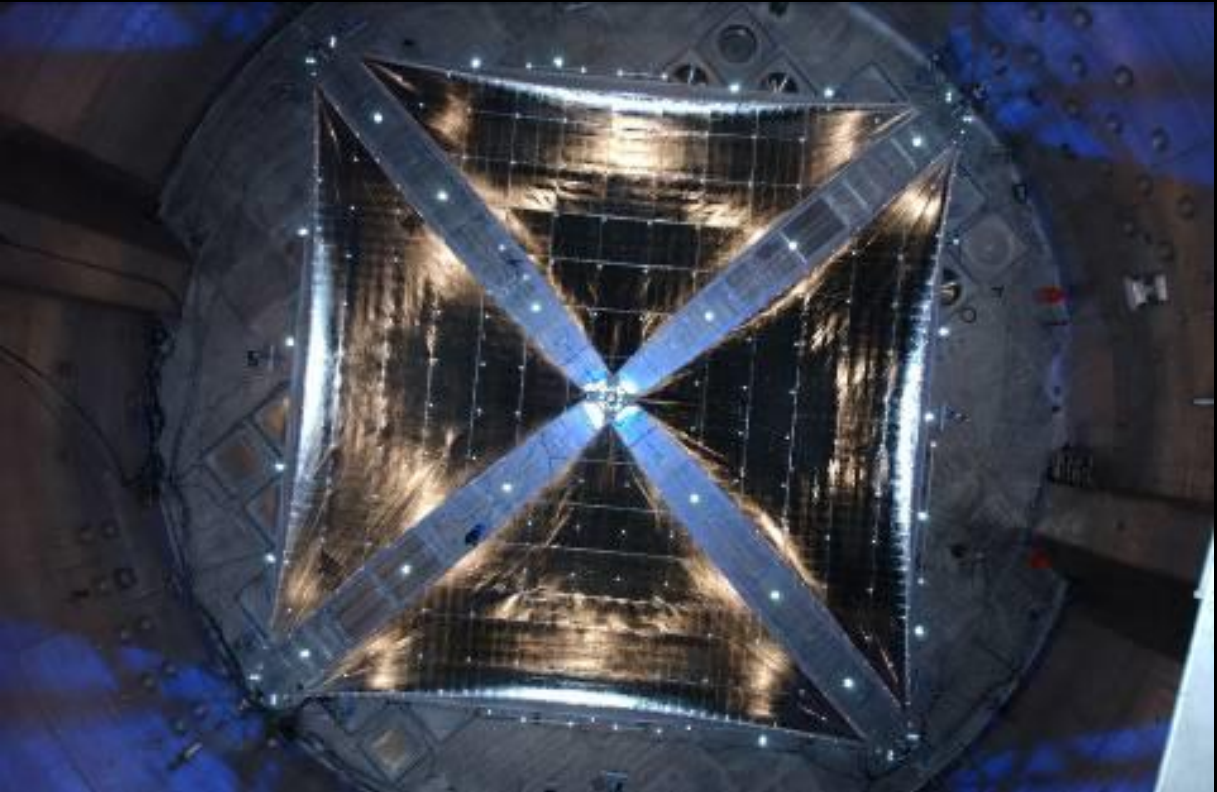


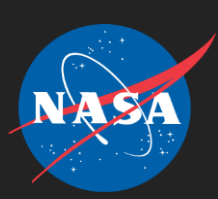
Rocket Failed





NASA Ground Tested Solar Sails in the Mid-2000's

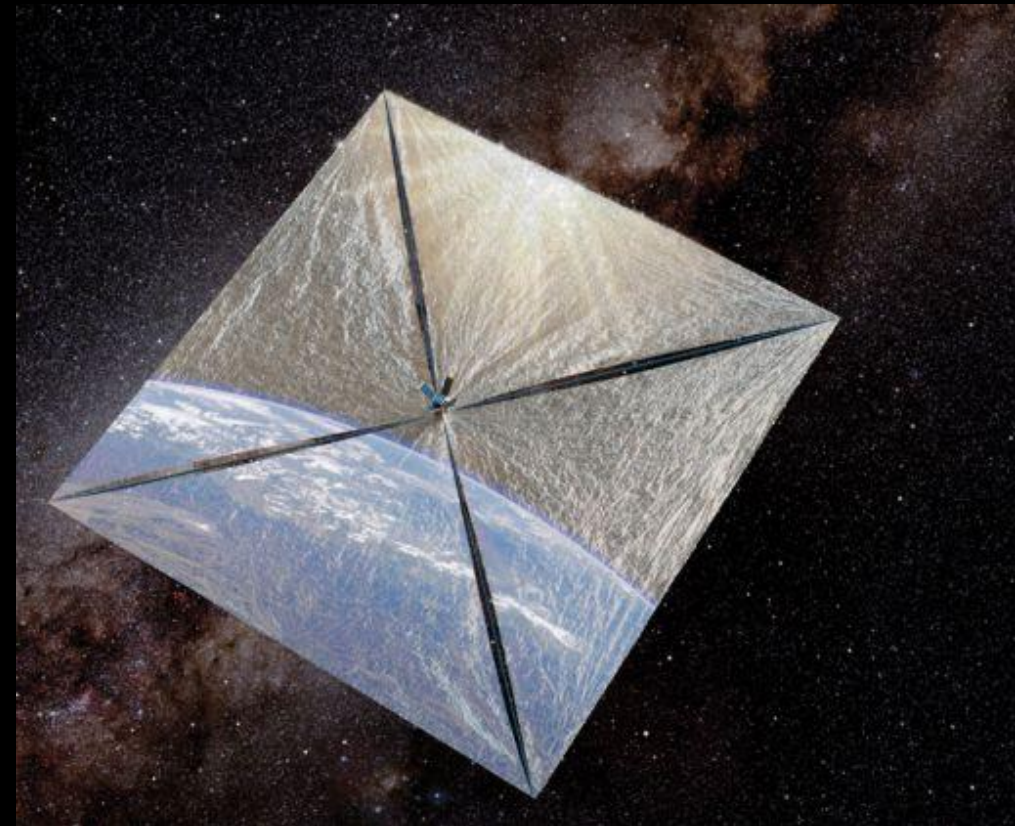


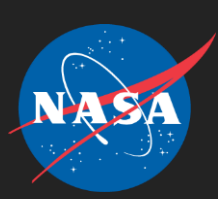


Nasa space technology Demo (2009)



Planned to be a space flight demonstration of the solar sail developed and tested as part of the ground sail test program



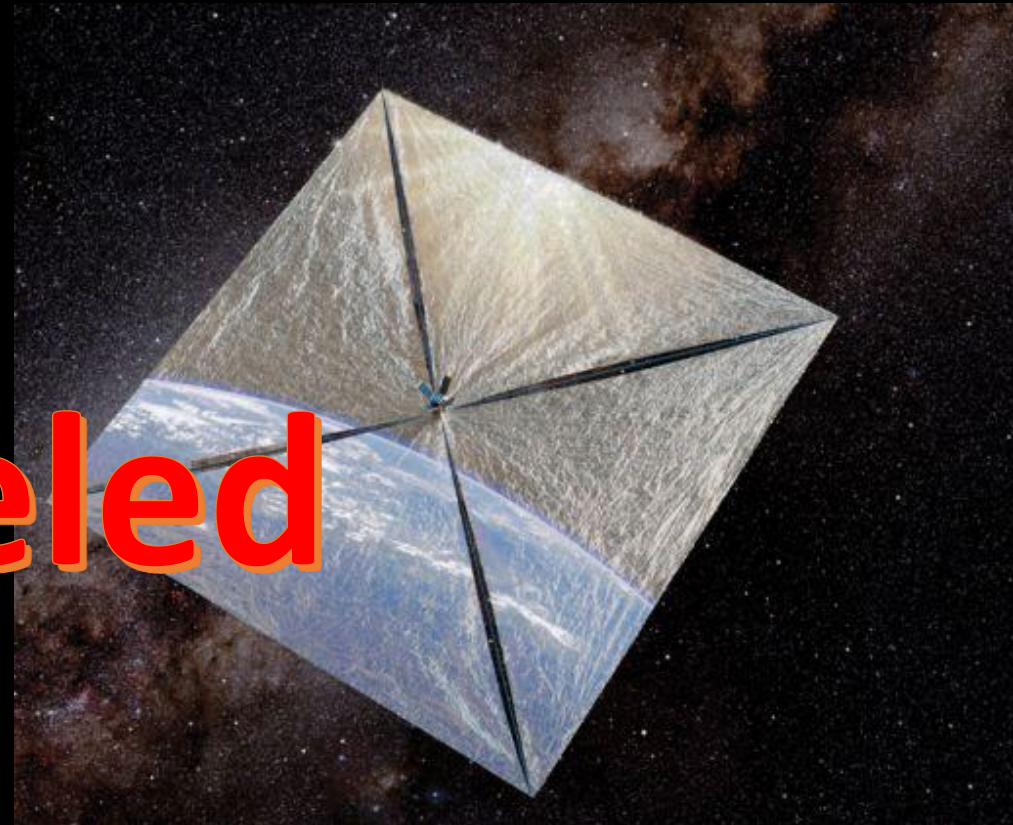


NASA space technology Demo (2009)



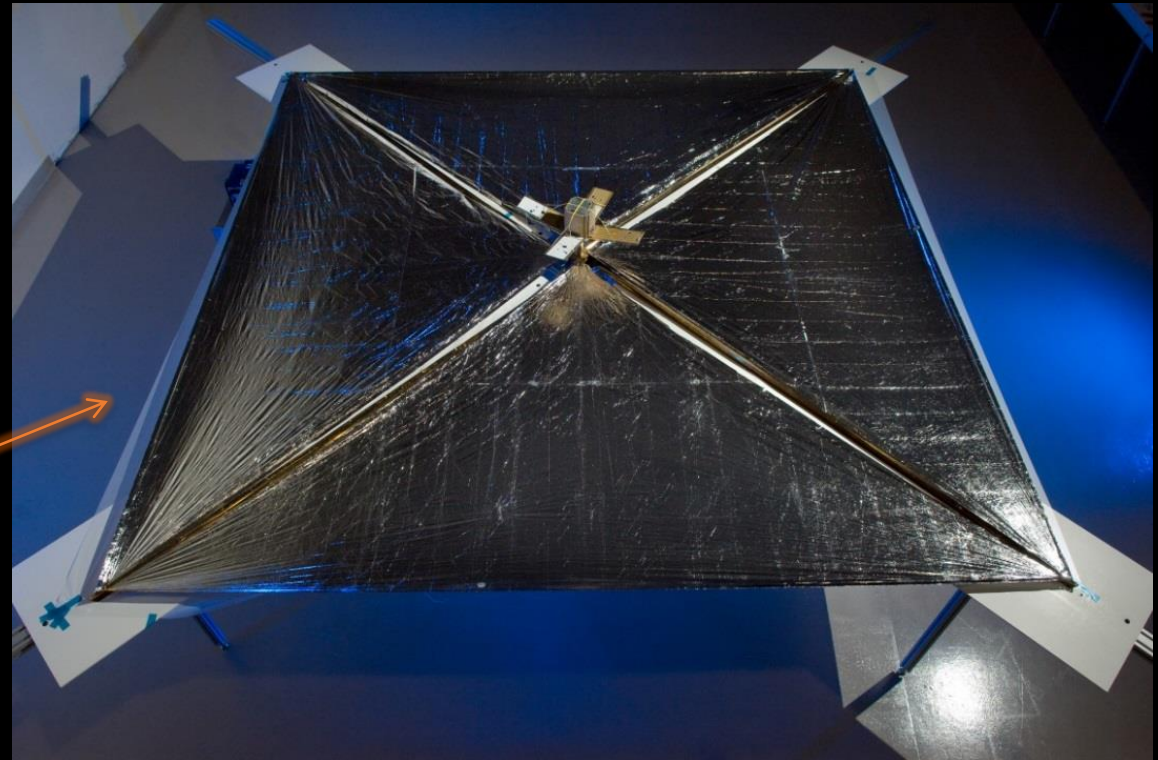
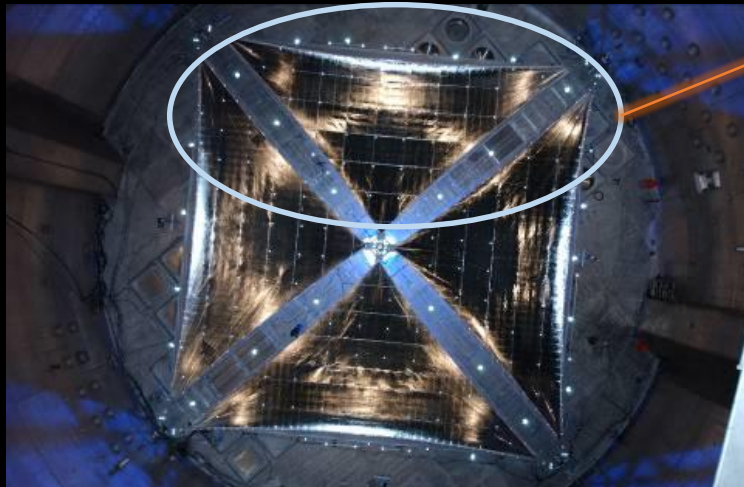
Planned to be a space flight demonstration of the solar sail developed and tested as part of the ground sail test program

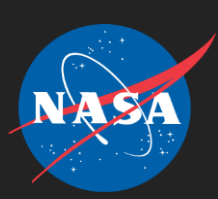
Cancelled



Mission Description:

- 10 m² sail
- Made from tested ground demonstrator hardware



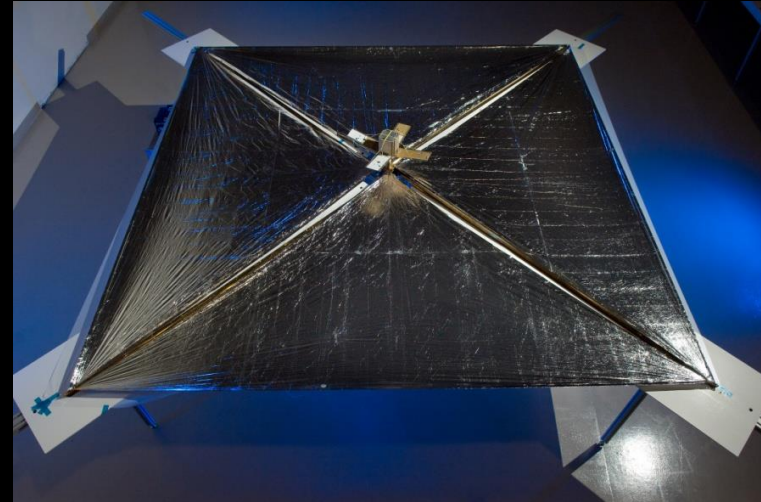


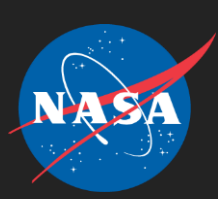
NanoSail-D1 Flight (2008)



Launch:

- Falcon-1, flight 3
- Kwajalein, Missile Range
- Primary payload: Air Force PnPSat



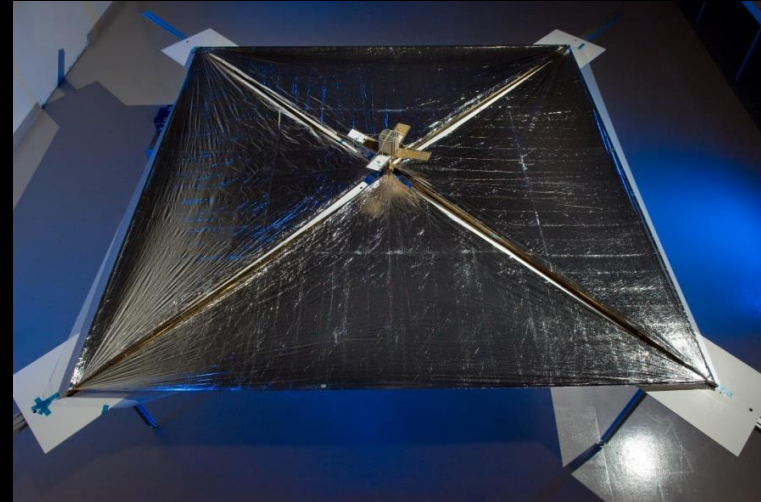


NanoSail-D1 Flight (2008)



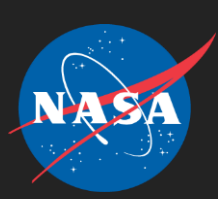
Launch:

- Falcon-1, flight 3
- Kwajalein, Missile Range
- Primary payload: AFRL PnPSat
- Secondary P-POD payloads (2)



Rocket Failed





NanoSail-D2 Mission Configuration (2010)



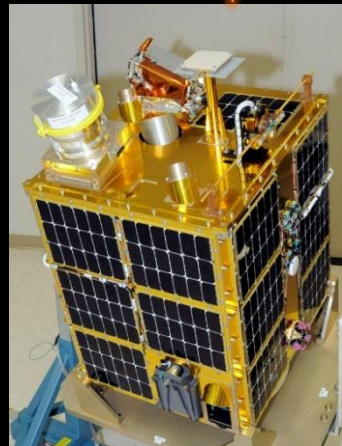
3U CubeSat: 10 cm X 10 cm X 34 cm

- Deployed CP-1 sail: 10 m² Sail Area (3.16 m side length)
- 2.2 m Elgiloy Trac Booms
- UHF and S-Band communications

Nanosail-D2 in Orbit August 19 2011 01h 19m 28s UT
 Clay Center Observatory at Dexter and Southfield Schools
 42.307404N, -71.13722W (WGS84)
 www.claycenter.org Focal length: 12,200mm,
 Aperture = 640mm Ritchey-Chretien
 Contact: Ron Dantowitz (rondantowitz@gmail.com)

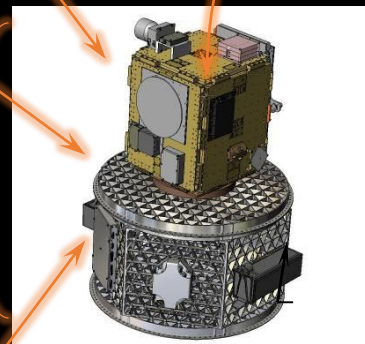


AFRL Satellite (Trailblazer)



HSV-1

Adapter

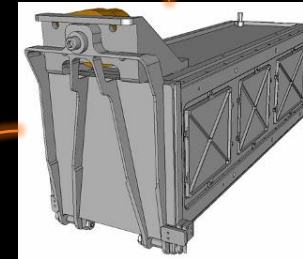


PreSat (ARC)

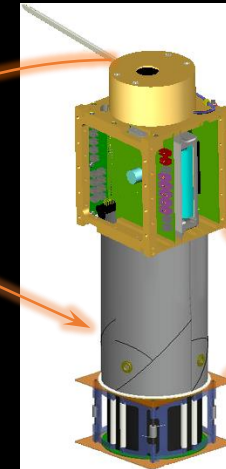
NanoSail-D (MSFC)

Ride Share Adapter (Space Access Technology)

Boom & Sail Spool (ManTech SRS)



PPOD Deployer (Cal-Poly)



NanoSail-D (Aluminum Closeout Panels Not Shown)

Spacecraft Bus (Ames Research Center)

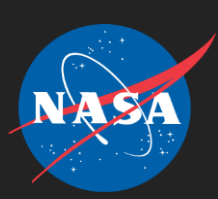
Bus interfaces Actuation Electronics (MSFC/UAH)

Stowed Configuration

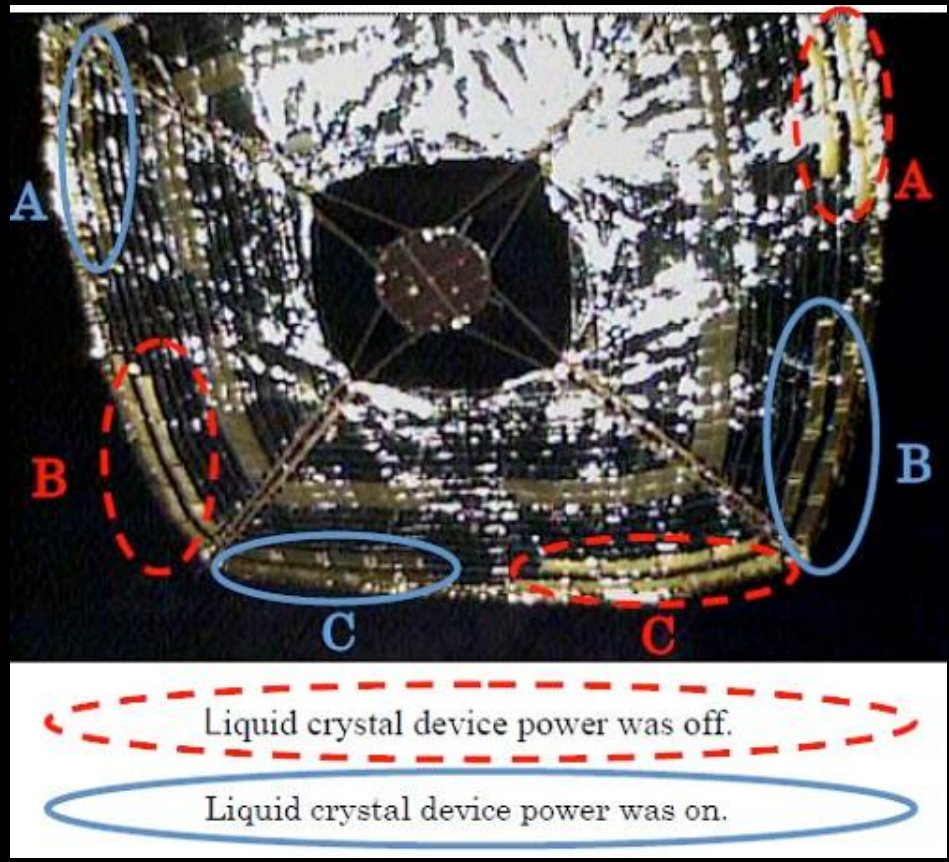


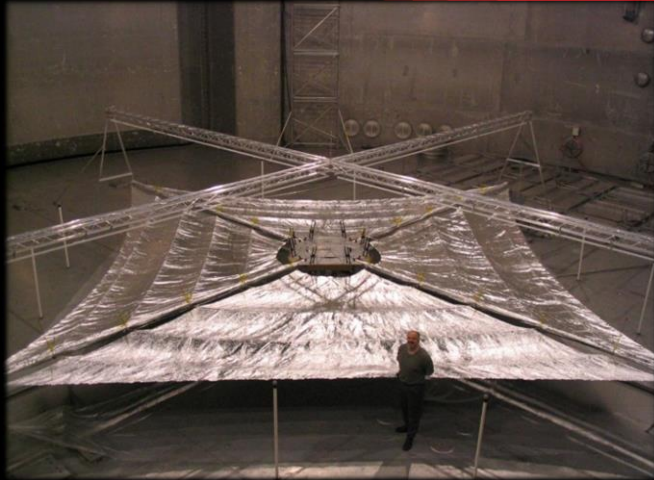
NSD-002

NSD-001



Interplanetary Kite-craft Accelerated by Radiation of the Sun (IKAROS)





83 m² ISP L'Garde Solar Sail
2004



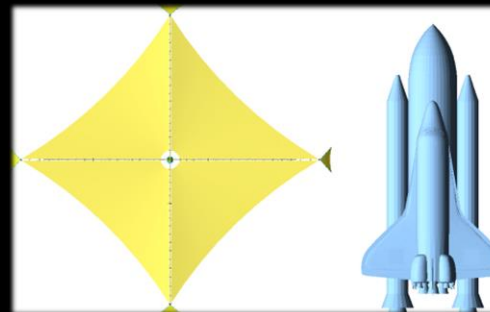
318 m² ISP L'Garde Solar Sail
2005

Design Heritage:

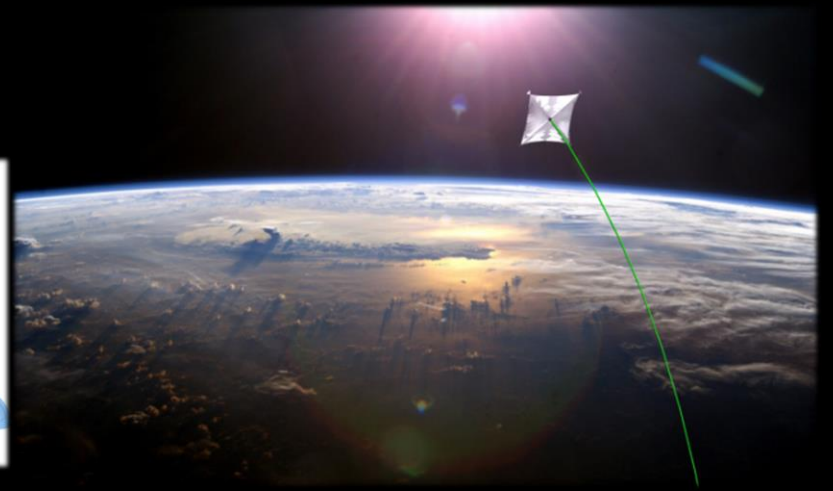
- Cold Rigidization Boom Technology
- Distributed Load Design
- Aluminized Sun Side
- High Emissivity Eclipse Surface
- Beam Tip Vane Control
- Spreader System Design

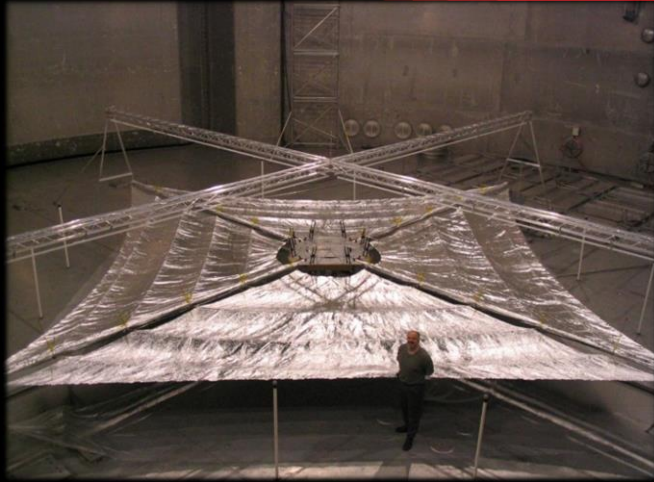
Design Features:

- High density packagability
- Controlled linear deployment
- Structural scalability
- Propellantless operation
- Meets current needs



1200 m² L'Garde Sunjammer Launch
2015





83 m² ISP L'Garde Solar Sail
2004



318 m² ISP L'Garde Solar Sail
2015

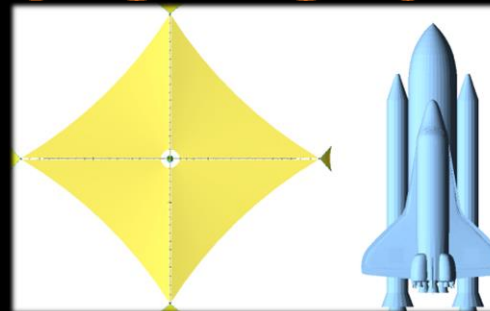
Design Heritage:

- Cold Rigidization Boom Technology
- Distributed Load Design
- Aluminized Sun Side
- High Emissivity Eclipse Surface
- Beam Tip Vane Control
- Spreader System Design

Canceled

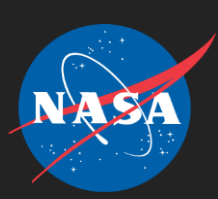
Design Features:

- High density packagability
- Controlled linear deployment
- Structural scalability
- Propellantless operation
- Meets current needs

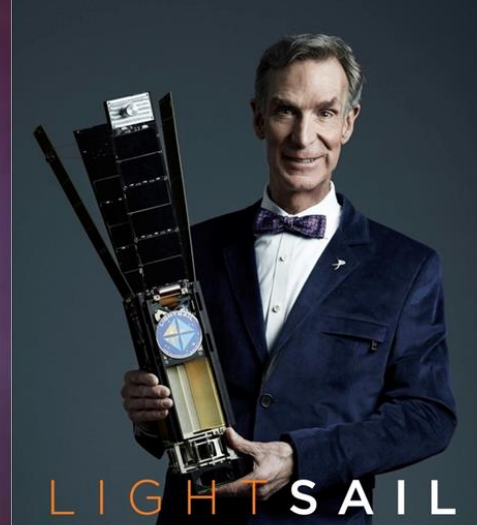
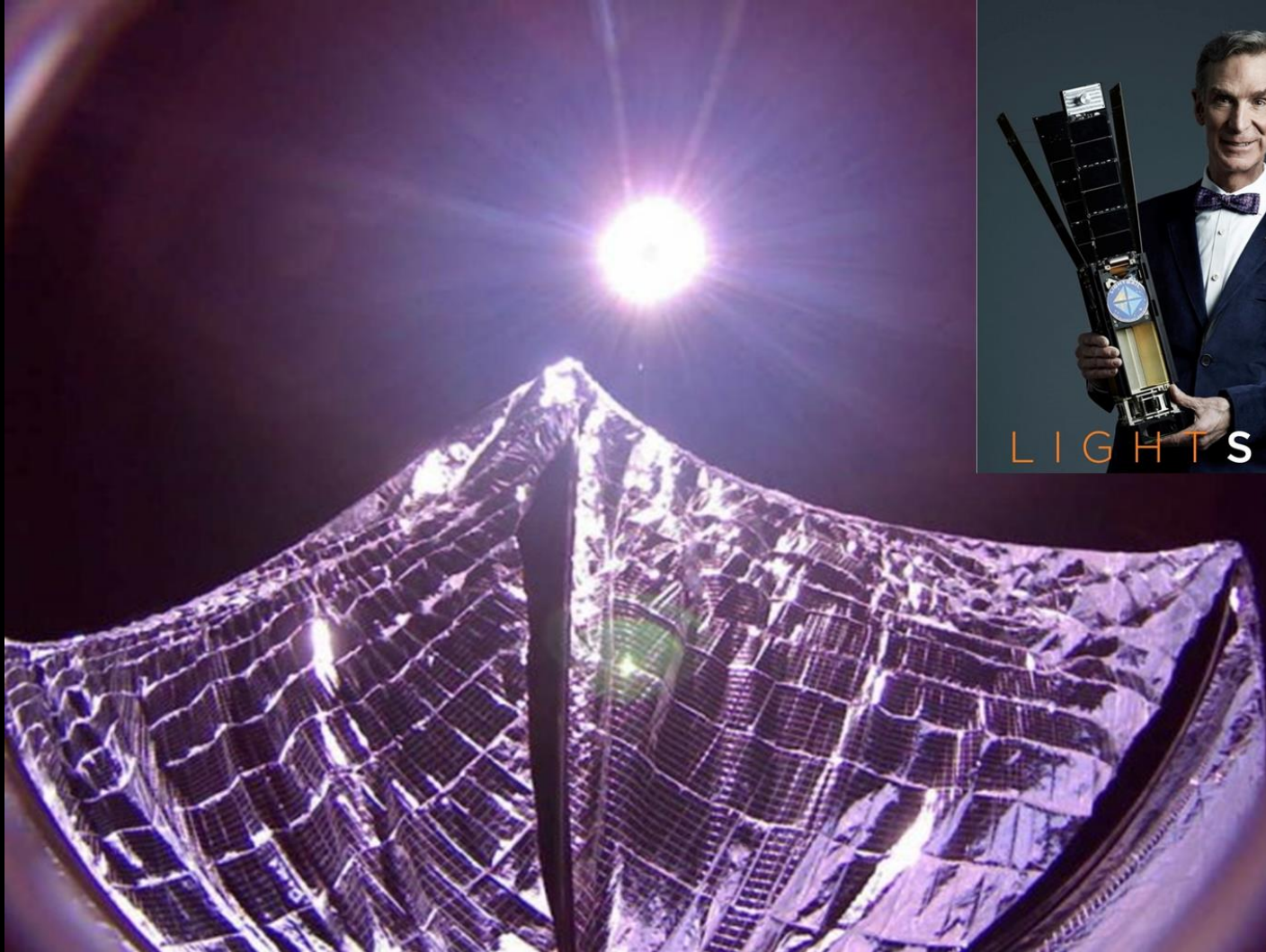


1200 m² L'Garde Sunjammer Launch
2015

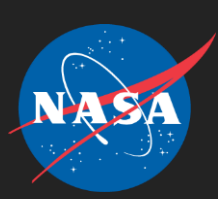




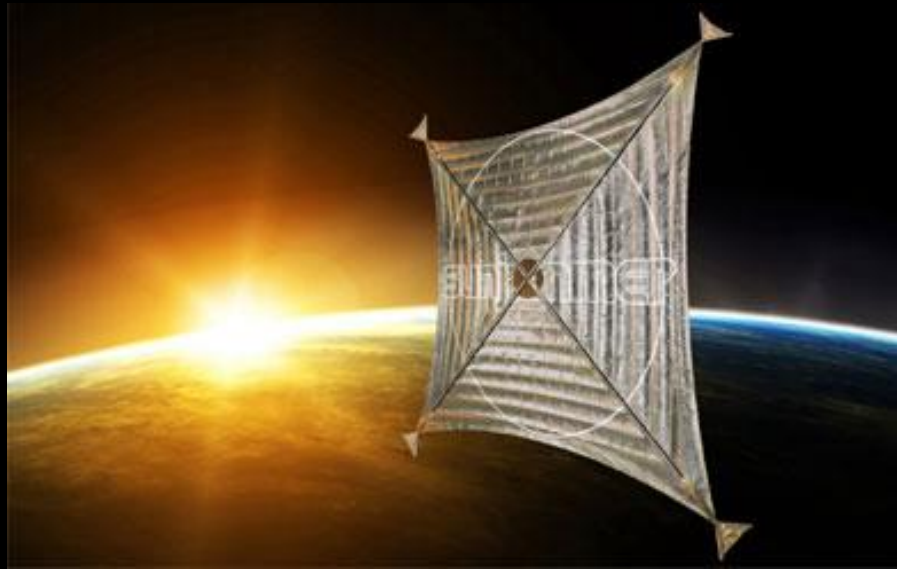
Lightsail-A (The planetary society)



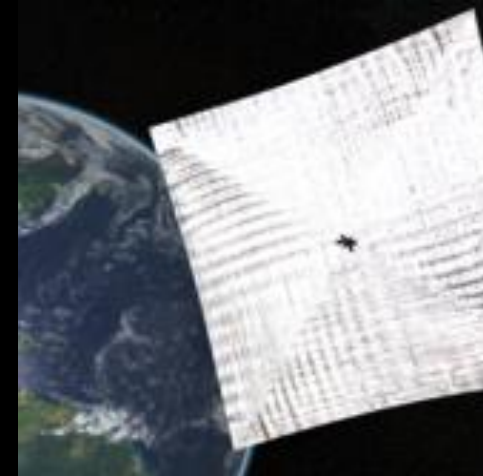
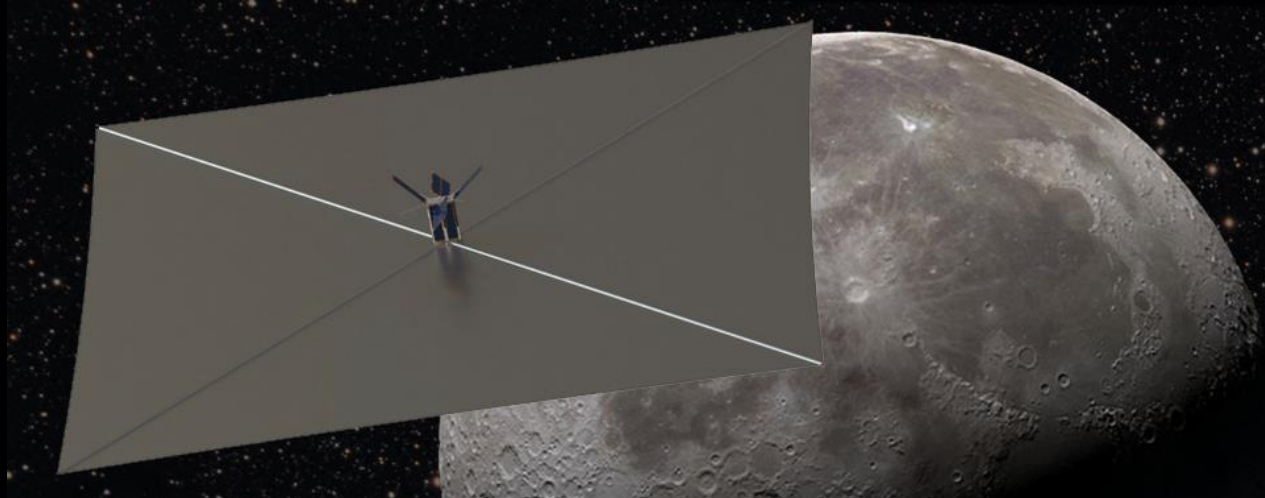
32 m²
No active 'sailing'
3U cubesat

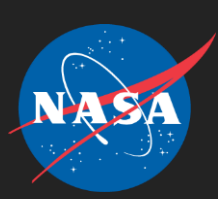


Solar Sails TODAY – Many Missions Planned



- NASA's *NEA Scout*
- The Planetary Society's *LightSail-B*
- The University of Surrey's *CubeSail*, *DeorbitSail*, and *InflateSail*
- University of Illinois' *CubeSail*





Near Earth Asteroid Scout



The Near Earth Asteroid Scout Will

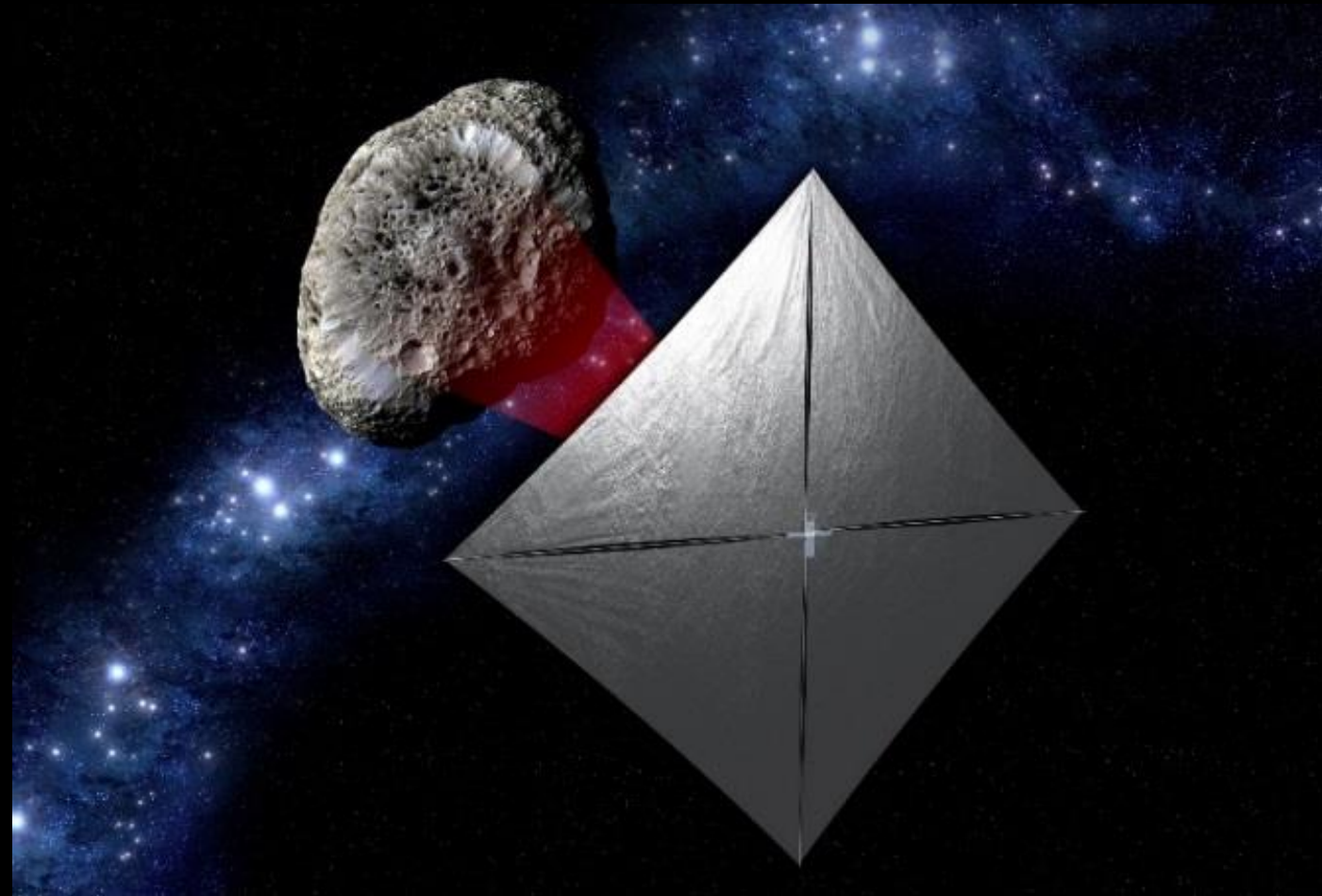
- Image/characterize a NEA during a slow flyby
- Demonstrate a low cost asteroid reconnaissance capability

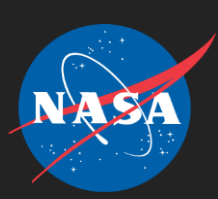
Key Spacecraft & Mission Parameters

- 6U cubesat (20 cm X 10 cm X 30 cm)
- ~86 m² solar sail propulsion system
- Manifested for launch on the Space Launch System (EM-1/2017)
- Up to 2.5 year mission duration
- 1 AU maximum distance from Earth

Solar Sail Propulsion System Characteristics

- ~ 7.3 m Trac booms
- 2.5 μ aluminized CP-1 substrate
- > 90% reflectivity





Flight System Overview



Mission Concept

Characterize a Near Earth Asteroid with an optical instrument during a close, slow flyby

Payload

Malin Space Science Systems ECAM-M50 imager

Mechanical & Structure

"6U" CubeSat form factor (~10x20x30 cm)

Propulsion

CP-1 solar sail (based on NanoSail-D2)

Avionics

Radiation tolerant LEON3-FT architecture

Electrical Power System

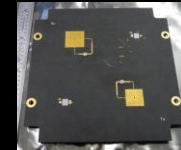
Simple deployable solar arrays

Telecom

JPL Iris
2 pairs of INSPIRE-heritage LGAs (RX/TX)
8x8 element microstrip array HGA (TX)



Rad Tolerant Avionics (JPL)



LGA (JPL)



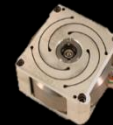
Star Tracker (Blue Canyon)



NEA Imager (Malin)



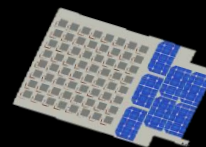
Iris 2.0 Transponder (JPL)



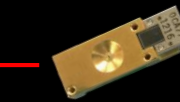
RWA (Blue Canyon)



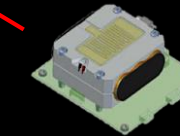
Solar Sail - Stowed (MSFC)



Solar Panels & HGA (MMA/AntDevCo)



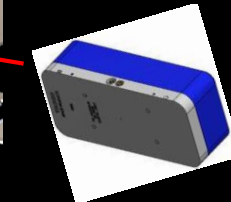
Coarse Sun Sensors (SSBV)



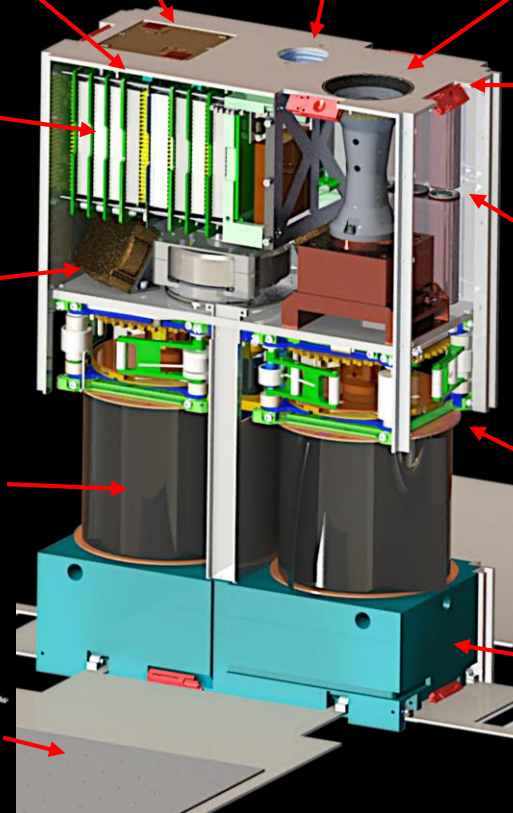
18650 Lithium Batteries (SDL/Panasonic)

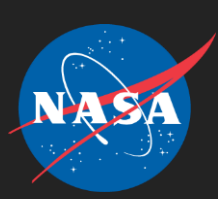


TRAC Boom Assembly (MSFC)



RCS (VACCO)

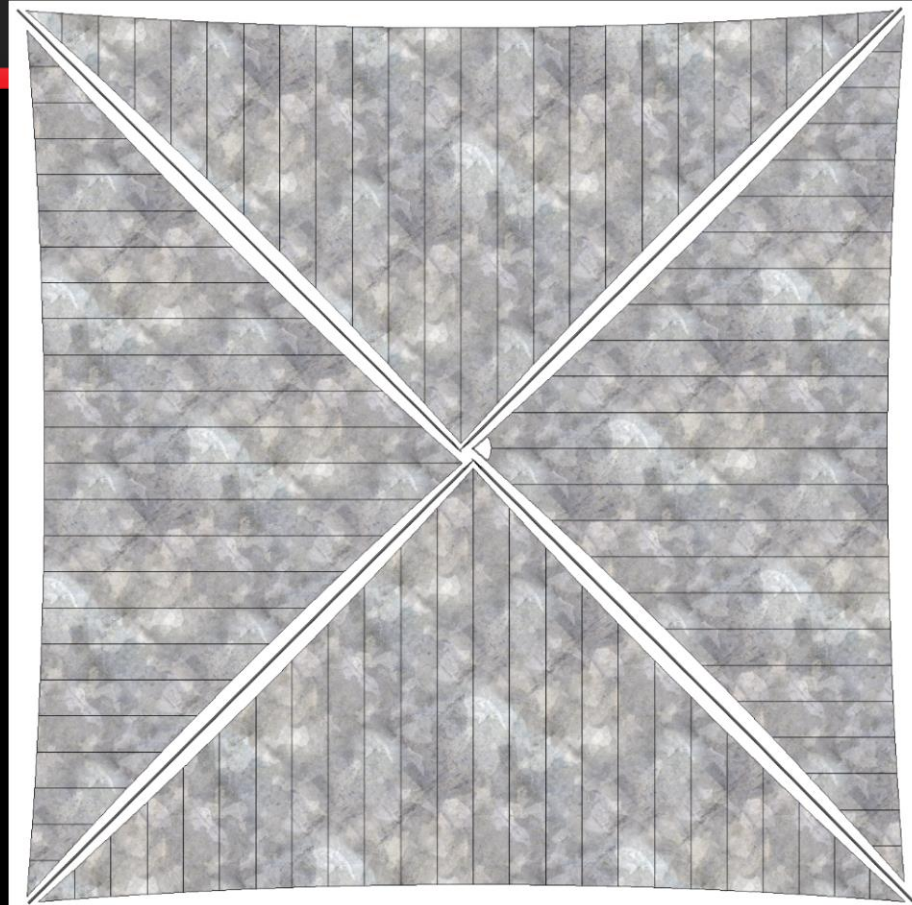




NEA Scout Approximate Scale



Deployed Solar Sail



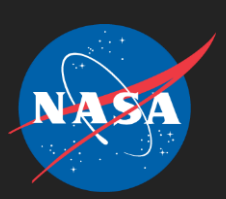
School Bus



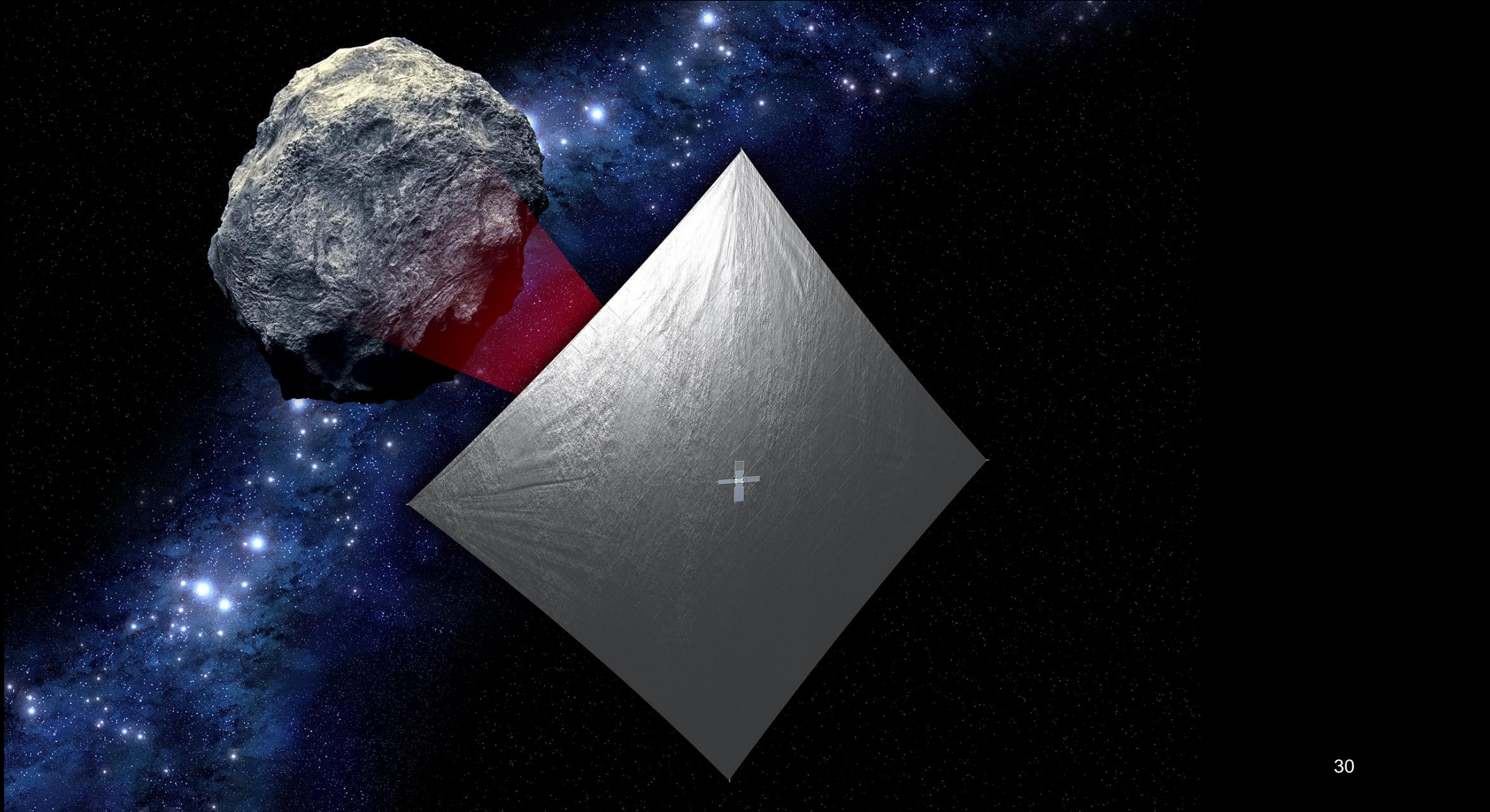
6U Stowed Flight System

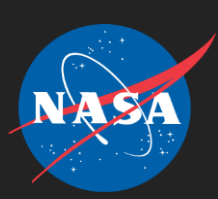


Folded, spooled and packaged in here



Near Earth Asteroid (NEA) Scout





University of Surrey's InflateSail



- **InflateSail** is an inflatable, rigidizable sail for flight in Low Earth Orbit:
 - 3U CubeSat with deployed sail area of 10 m²
 - Sail supported by bistable booms
 - Inflation is driven by Cool Gas Generators (CGG): low system mass, long lifespan

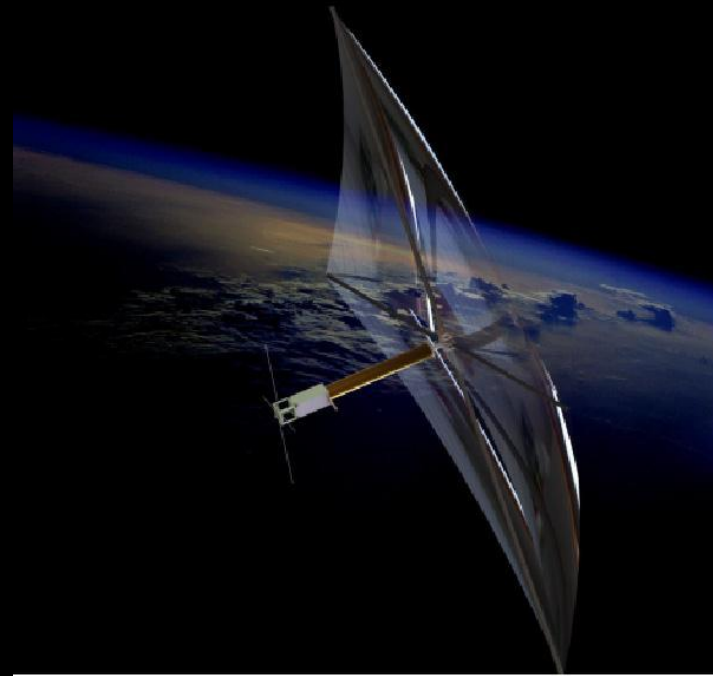
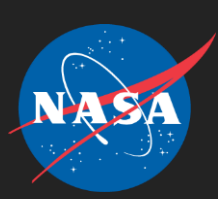


Fig. 1: InflateSail design concept



Fig. 2: 80 mg CGG George C. Marshall Space Flight Center

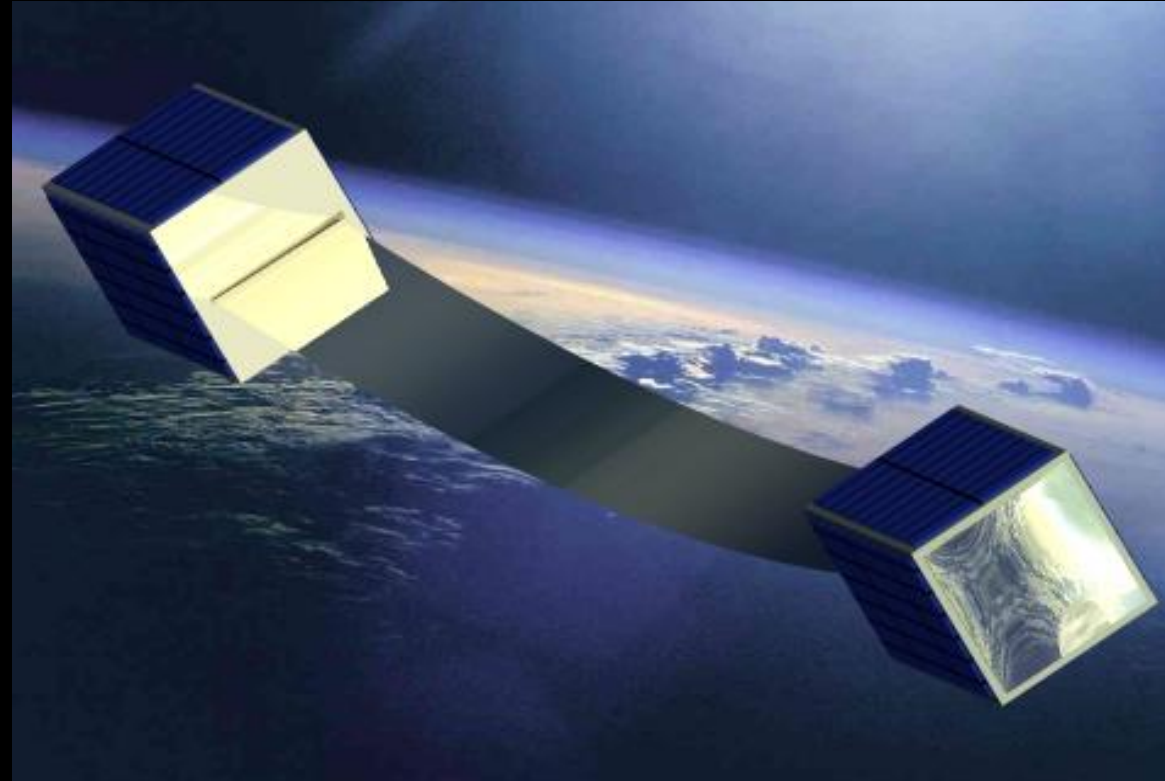


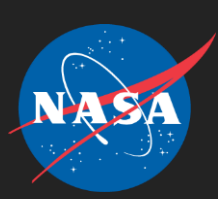


Cubesail CubeSat Solar Sail Propulsion Demonstration

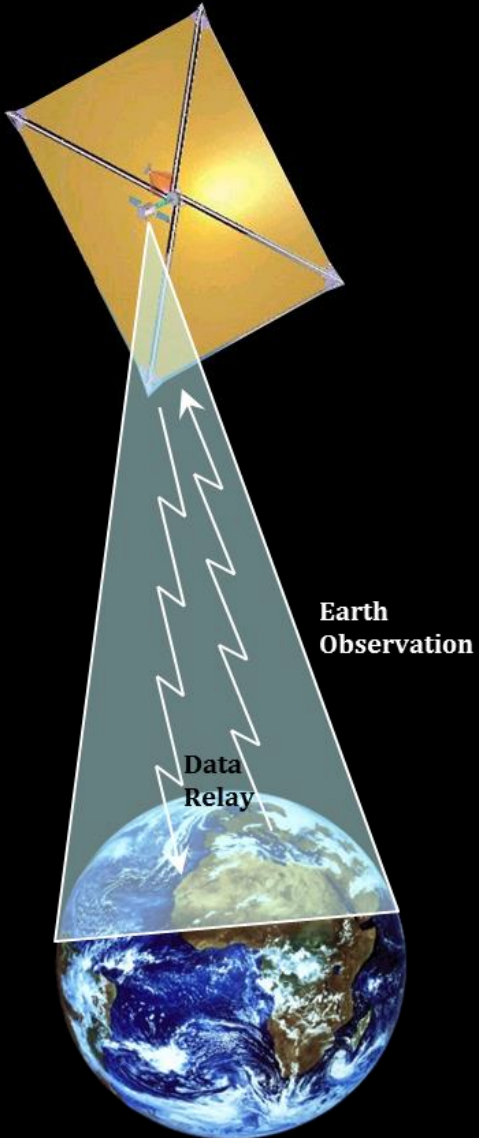


- The University of Illinois at Urbana-Champaign (UIUC), working with NASA MSFC, NSF, and CU Aerospace, built the flight hardware for a CubeSat-based 20 m² solar sail orbit raising demonstration mission
- Selected for flight under the NASA CubeSat Launch Initiative



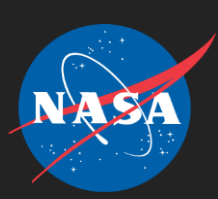


Continuous Polar Observations



Sailcraft over the polar regions of the Earth

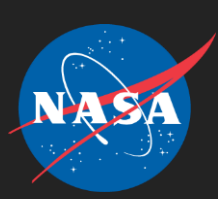
Sail tilted so the light pressure from the sunlight reflecting from it is exactly equal and opposite to the gravity pull of the Earth.



Deploy a large ($>10,000 \text{ m}^2$) solar sail near the sun to enable travel 5X faster than Voyager



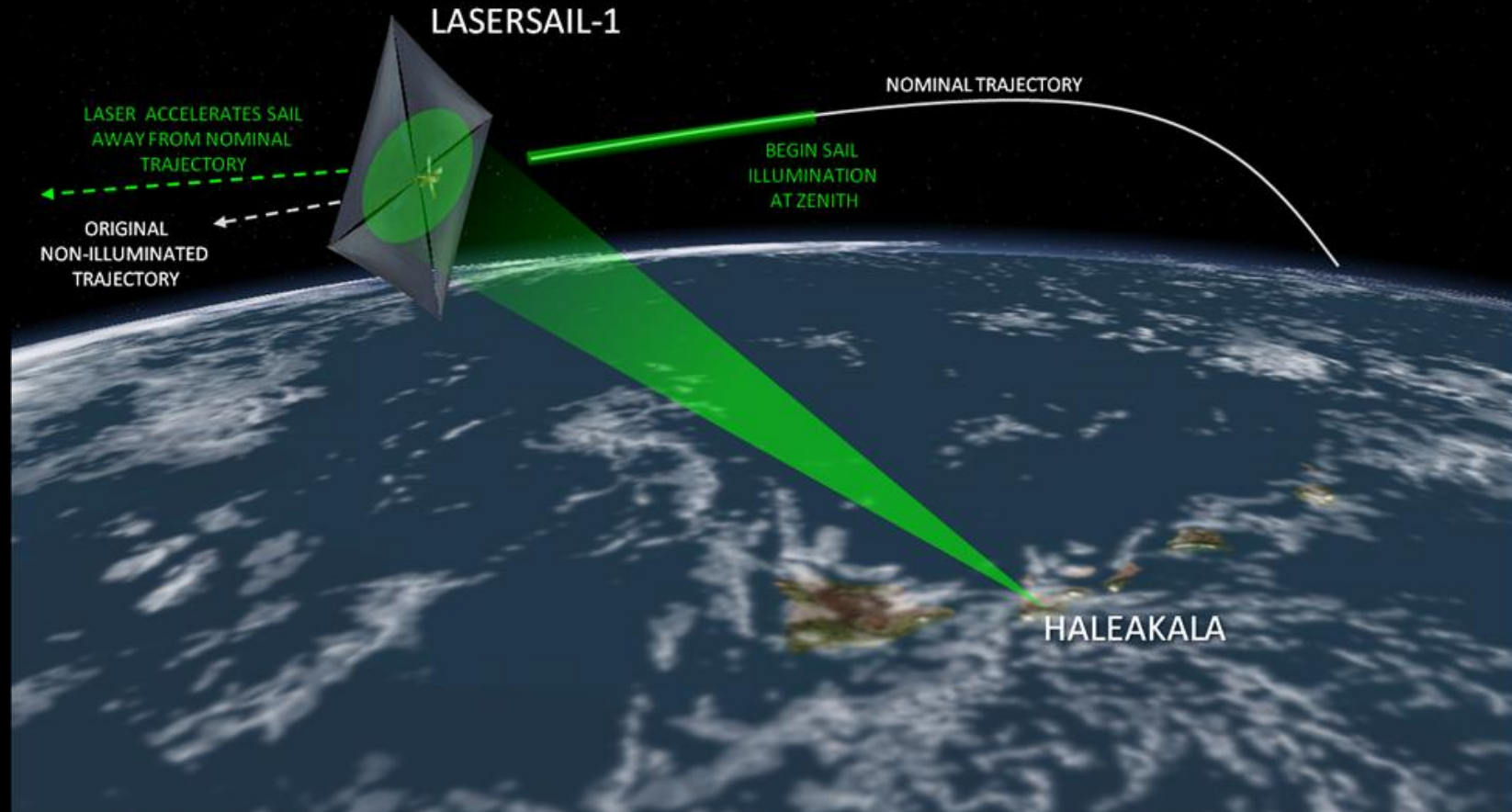
Goal: Reach 250 Astronomical Units within 20 years of launch

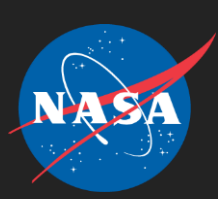


Laser Sailing: The Next Big Step






Ground to space laser illumination of a solar sail to impart measurable ΔV

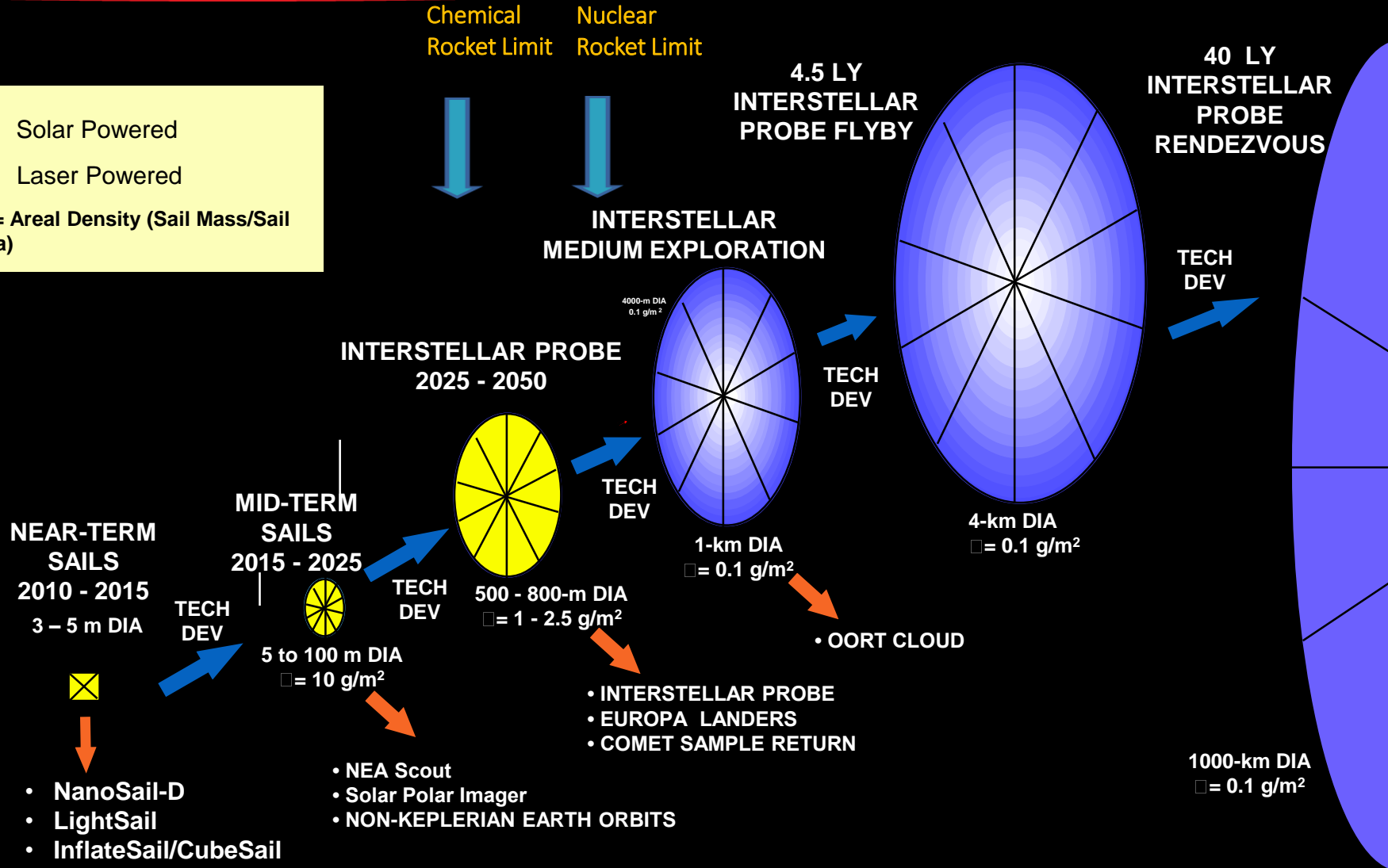


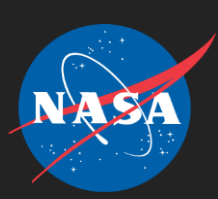


My Real Motive...

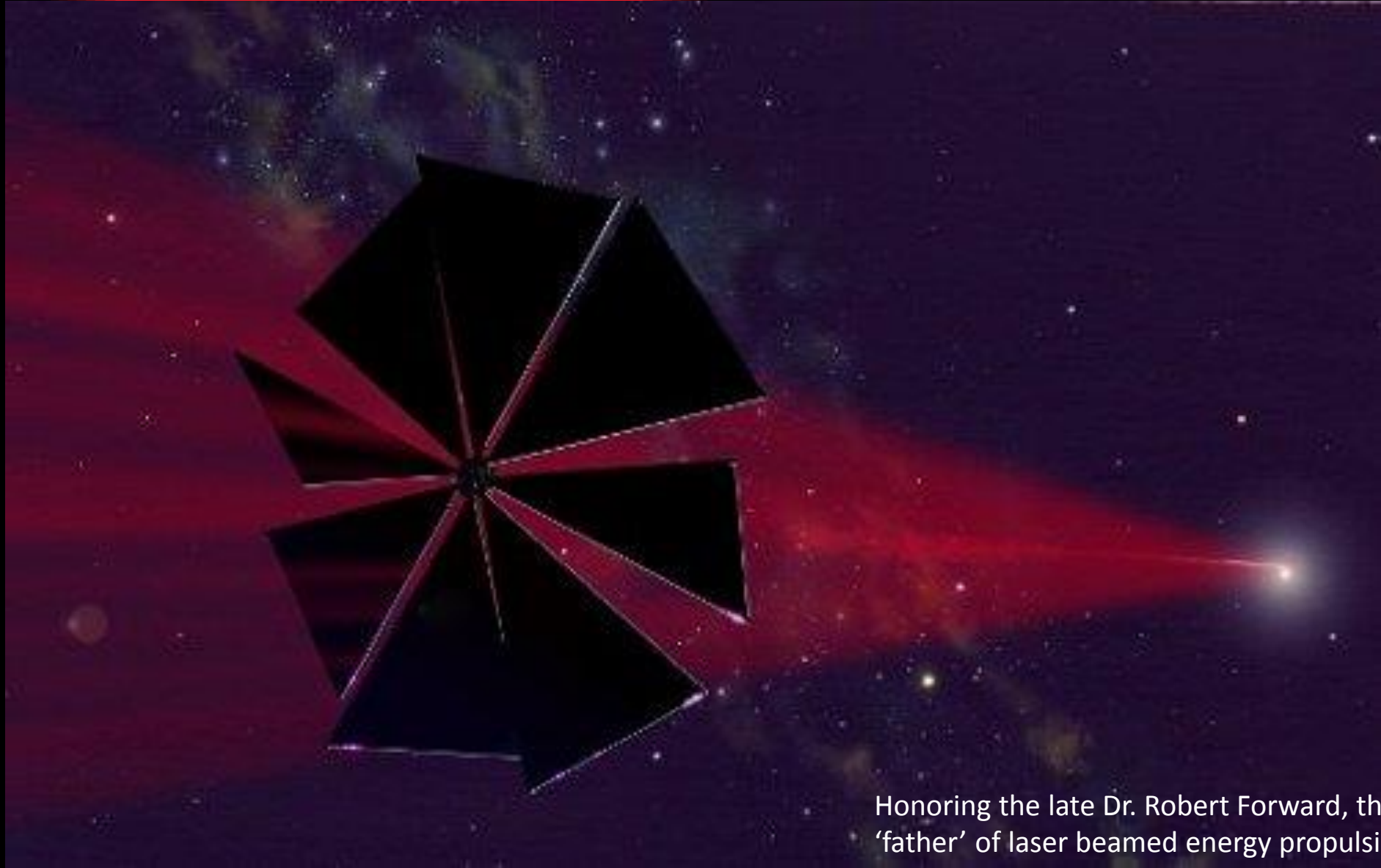


 Solar Powered
 Laser Powered
 = Areal Density (Sail Mass/Sail Area)





Solar Sails: A Step Toward the Stars



Honoring the late Dr. Robert Forward, the
'father' of laser beamed energy propulsion

